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# 1. RflySim Automatic Code Generation Simulink Configura tion Interface

#### 1.1. The Basic Architecture of RflySim's Automatic Code Generation

#### System

The Pixhawk Pilot Support Package (PSP), also known as "Autopilot Support Package" in Ch inese, is an official toolbox released by MathWorks for Pixhawk. This toolbox allows the automati c compilation and deployment of Simulink models containing autopilot algorithms to Pixhawk har dware systems using the Embedded Coder in Simulink. Its main functionalities include:

- Capable of simulating and testing various aircraft models and autopilot algorithms within Sim ulink, with automatic deployment of algorithms to the autopilot.
- The toolbox provides practical examples including light control, remote controller data proce ssing, and attitude controllers.
- Numerous interface modules are provided in the toolbox for accessing both the software and hardware components of the autopilot.
- > Automatic logging of flight data from sensors, actuators, and deployed controllers.
- Ability to subscribe to and publish uORB topic messages. All data within the PX4 autopilot s oftware is temporarily stored in a uORB message pool. Subscribing to uORB allows reading t opics of interest from this pool, while publishing to uORB allows specific topics to be publish ed into the pool for use by other modules.

The PX4 software system can be divided into several small modules, each running independe ntly (in parallel threads). These modules transmit and interact with data through the subscription a nd publication functions of the uORB messaging module. After deploying the code generated by S imulink to the PX4 autopilot software, it does not affect the operation of the native PX4 autopilot s oftware. Instead, it adds an independent module called "px4\_simulink\_app" (running in a separate thread) parallel to the other modules. Since the native PX4 control algorithms may need to access t he same hardware output resources as "px4\_simulink\_app," this can lead to read-write conflicts. T herefore, the platform's one-click deployment script provides an option to automatically block the PX4 native firmware from accessing the actuators, ensuring that only the "px4\_simulink\_app" mo dule can output motor controls, as shown in the diagram below.



The PSP toolbox generates C code from control algorithms designed in Simulink. This code i s then imported into the source code of the PX4 software system to create a standalone program ca lled "px4\_simulink\_app" that runs independently. The toolbox invokes the compilation tool to com pile all the code into a PX4 software firmware file with the extension ".px4". This firmware file is downloaded and flashed into the flight controller, enabling the flight controller to execute the PX4 software with the generated algorithm code.

#### **1.2.** Internal Communication in the PX4 Software System

PX4 consists of two main components: the flight control stack, which primarily includes state estimation and flight control systems, and the middleware, a universal robotic application layer ca pable of supporting any type of autonomous robot. The middleware is responsible for internal/exte rnal communication and hardware integration.

All supported PX4 drone models, including other platforms like unmanned boats, cars, under water vehicles, etc., share a common codebase. The entire system employs a reactive design, mean ing:

- All functionalities can be divided into several replaceable and reusable components.
- Communication is done through asynchronous message passing.
- The system can handle different workloads.
  - The PX4 system achieves inter-module communication through a publish-subscribe mes sage bus called uORB. Utilizing this publish-subscribe message bus implies:
- The system is reactive; it operates asynchronously, updating immediately upon the arriva l of new data.
- All activities and communications within the system are fully parallelized.
- System components can access data from anywhere while ensuring thread safety.

uORB (Micro Object Request Broker, Micro Object Request Broker) is a crucial module in th e PX4/Pixhawk system, tasked with the entire system's data transmission. All sensor data, GPS, PP M signals, etc., are obtained from the chip and transmitted through uORB to various modules for c omputation and processing. In reality, uORB is a set of inter-process IPC communication modules. In Pixhawk, all functionalities are independently implemented and operate as process modules. Th e inter-process data exchange is crucial and must meet the real-time and ordered characteristics.

Internally, the flight controller utilizes the NuttX real-time ARM system. For NuttX, uORB is just a regular file device object that supports Open, Close, Read, Write, Ioctl, and Poll mechanism s. Through the implementation of these interfaces, uORB provides a "point-to-multipoint" inter-pr ocess broadcast communication mechanism. Here, "point" refers to the communication message's "source," and "multipoint" means a source can have multiple users to receive and process data. The relationship between the "source" and "user" is such that the source does not need to consider whe ther a user can receive a broadcast message or when they receive it. It simply pushes the data to th e uORB message "bus." For the user, the number of times the source pushes messages is not impor tant; what matters is retrieving the latest message. In the communication process, the sender is onl y responsible for sending data without caring about who receives it or whether the receiver can receive all the data. Similarly, the receiver does not care who sent the data or whether all data was rec eived during the process.

uORB does not guarantee that all sender data can be received by the receiver during data publ ishing and reception; it only ensures that the receiver can receive the latest data when desired. The separation of sending and receiving allows modules during flight to operate independently without interference. In practice, a uORB can be published by multiple senders and received by multiple r eceivers, forming a many-to-many relationship. Publishers update and publish data to the uORB pl atform at a certain frequency without concern for who is receiving. Subscribers can retrieve data at any time. For further learning resources, please refer to: <u>https://docs.px4.io/main/zh/middleware/u</u> orb.html



#### **1.3. Simulink Configuration Interface**

The Embedded Coder module of MATLAB/Simulink can generate readable, compact, and fas t C and C++ code for use with embedded processors in large-scale production. It extends the capab ilities of MATLAB Coder and Simulink Coder, providing precise control over generated functions, files, and data through advanced optimizations. These optimizations enhance code efficiency and f

acilitate integration with existing code, data types, and calibration parameters. Third-party develop ment tools can be integrated to build executable files for deployment on embedded systems or rapi d prototype boards.

To generate code for Pixhawk series flight controller hardware after completing the model in Simulink, follow these settings in the Model Configuration Parameters:

Pa Exp3_BlankTemp - Simulink			- 0
File Edit View Display Diagram Simulation Analysis Code Tools	Help		
🗞 • 🔄 • 拱 🤄 💠 🛧 🔡 🏟 • 📾 • 🕪 😓 🕨	Normal	• 🧭 • 🛗 •	
Exp3_BlankTemp  Model Configuration Parameter	s Ctrl+E		
Exp3_BlankTemp     Simulation Target For MATLAB &	Stateflow		
Model Properties			
e.			

1) Set the solver type to: Fixed-step, Solver selection: discrete. That is, select a discrete fix ed-step solver.

Q Search			
Solver Data ImportExport • Optimization • Diagnostics Hardware Implementation Model Referencing Simulation Target • Code Generation • Coverage • HDL Code Generation	Simulation time Start time: 0.0 Solver options Type: Fixed-step    Additional parameters   Kixed-step size (fundamental sample time): aut Tasking and sample time constraint: Unconstrained Treat each discrete rate as a separate task Allow tasks to execute concurrently on targe Allow tasks to	Stop time: inf  Solver: discrete (no continuous states)  to  t t transfer isible iority	

2) Go to Hardware Implementation, set the Hardware board to Pixhawk PX4, choose the o perating system to be Nuttx, which is supported by the PX4 software system. Set the clo ck frequency to 250, and for Target hardware resource, select "build only."

Solver	Hardware board: Pixhawk PX4			
Data Import/Export	Code Generation system target file: ert.tlc			
Optimization	Device vendor: ARM Compatible		Device type: ARM Cortex	·
Signals and Parameters			Denice type:	
	Device details			
Hardware Implementation	Hardwara board pottings			
Model Referencing	Hardware board settings			
Simulation Target	<ul> <li>Operating system/scheduler settings</li> </ul>			
Code Generation	Operating system options			
Coverage				
HDL Code Generation	Operating system: NuttX			-
	Base rate task priority: 250			
	Target hardware resources			
	Groups			
	Build options	Build action: Build		
	Clocking	Duild action. Duild		
	External Mode Options			
	Uploading Options (Windows-only)			
	Hard Real-Time constraints			

3) Under Code Generation, set the System target file to ert.tlc from Embedded Coder, and c hoose the C language. Select the Pixhawk Toolchain as the toolchain, and configure it fo r Faster Builds.

Configuration Parameters: Exp3_BlankTemp/Configuration (Active)								
Q Search								
Solver Data Import/Export ▼ Optimization Signals and Parameters Stateflow Diagnostics	Target selection       System target file:       Language:       C       Description:     Embedded Coder	Br	owse					
Hardware Implementation Model Referencing Simulation Target ▼ Code Generation	Viagnosius     Hardware Implementation     Model Referencing     Simulation Target     ☐ Generate code only     Simulation Target     ☐ Package code and artifacts     Zip file name: <emply></emply>							
Report Comments Symbols Custom Code Interface Code Style Verification Templates	Toolchain settings         Toolchain:       Pixhawk Toolchain         Build configuration:       Faster Builds <ul> <li>Toolchain details</li> </ul>		•					
Code Placement Data Type Replacement Memory Sections Coverage HDL Code Generation	Code generation objectives Prioritized objectives: Unspecified Check model before generating code: Off	Set Objec Check Mc	otives odel					
	OK Cancel	Help	A	pply				

 In the Code Generation interface settings, set the code's dependency library to GCC AR M Cortex-A9, and configure the support settings as shown in the figure below.

Configuration Parameters: Exp3_B	lankTemp/Configuration (Active)			-	
Q Search					
Solver Data Import/Export > Optimization > Diagnostics Hardware Implementation Model Referencing Simulation Target > Code Generation	Software environment Code replacement library: Shared code placement: Support: floating-point n absolute time	GCC ARM Cortex-As Auto umbers	<ul> <li>✓ non-finite numbers</li> <li>✓ continuous time</li> </ul>	complex numbers	•
Report Comments Symbols Custom Code	Code interface packaging:	Nonreusable function	n data structure		•
Interface         Data exchange interface           Code Style         Generate C API for:					
Templates Code Placement Data Type Replacement Memory Sections	signals         ASAP2 interface         External mode	parameters	States	root-level I/O	
► HDL Code Generation					
				OK Cancel Help	Apply

For more options for automatic code generation settings, please refer to the document: <u>0.Ap</u> <u>iExps\3.DesignExps\Readme.pdf</u>

## **1.4.** Code Modification Interface

The one-click installation script for the RflySim platform makes some modifications to the official PX4 source code for better compatibility with the platform. The core modifications include:

1) In \*\PX4PSP\Firmware\boards, locate the corresponding compilation command cmake file, such as px4\_fmu-v6x\_default in \*\PX4PSP\Firmware\boards\px4\fmu-v6x\default.cmake, and ad d the module compilation registration for px4\_simulink\_app.

2) In \*\PX4PSP\Firmware\ROMFS\px4fmu\_common\init.d\rcS startup script, add the automa tic startup for px4\_simulink\_app.

3) In \*\PX4PSP\Firmware\src\modules directory, create a px4\_simulink\_app folder and gener ate source code that meets the compilation requirements.

4) Modify the source code under Firmware to disable PX4's own motor output, allowing px4\_ simulink\_app to gain control over the motors.

#### **1.5.** Custom Source Code Import Interface

The 2.FirmwareZip directory in the RflySim platform's installation package contains various PX4 source code firmware, and it supports importing custom-developed PX4 source code. When r edeploying firmware using the one-click installation script, it first deletes the \*\PX4PSP\Firmware folder. Then, based on the selected option, it unpacks "2.FirmwareZip\PX4Firmware\*\*\*.zip" to th e \*\PX4PSP directory. Finally, it extracts the contents of PX4Firmware\*\*\*Update.zip and forceful ly overrides them into the Firmware directory.

In PX4Firmware\*\*\*.zip, the official source code is stored, downloaded from GitHub without any modifications. PX4Firmware\*\*\*Update.zip contains the files we modified, which will overwri te the Firmware directory. Therefore, there are two ways to deploy custom source code:

- Directly package the modified Firmware directory, rename it according to your version a s PX4Firmware\*\*\*\*.zip (see naming rules in 2.FirmwareZip\readme.txt), and delete th e PX4Firmware\*\*\*\*Update.zip file. This way, the one-click installation script will use y our own script for deployment.
- Alternatively, you can directly place your modified source code parts, structured by File Contents, inside PX4Firmware\*\*\*Update.zip. During deployment, they will be copied i n and forcibly replace the original code.

The RflySim platform also supports other PX4 firmware versions, such as 1.9.2, 1.10.2, etc., as me ntioned in the 2.FirmwareZip\readme.txt file, as shown in the following figure.



# 2. Simulink/PSP Toolbox Module Interface

PSP3.04 provides some Simulink modules as the hardware interface to Pixhawk. These modu les are only responsible for generating the corresponding interface code and do not include the mo deling of the peripheral hardware. It is best to organize your own control system into a Simulink m odule at the time of simulation, leaving the necessary interfaces in order to connect with these hard ware interface modules, so that it is also easy to reuse. These hardware modules can be viewed in Simulink's toolbox Pixhawk Target Blocks, as shown in the figure below, which consists of four su blibraries: ADC and Serial, Miscellaneous Utility Blocks, Sensors and Actuators, and uORB Read and Write.



## 2.1. ADC and Serial — ADC and Serial Communication Library

As shown in the figure below, the ADC read module can obtain the data of 3 external ADC ch annels, and the serial port module can read and write the specified serial port.

Simulink Library Browser	_	×
💠 🔶 mavlink 🗸 🗸 💆 🕶 🔁 😨		
Pixhawk Target Blocks/ADC and Serial		
Image Acquisition Toolbox Instrument Control Toolbox > LIE HDL Toolbox Nodel Predictive Control Toolbox > Neural Network Toolbox > Phased Array System Toolbox > Powertrain Blockse Read ADC Channels Serial Miscellaneous Utility Blocks Sensors and Actuators uORB Read and Write > Powertrain Blockset Report Generator > RF Blockset > Simulink Control Toolbox SimUlink Decign > Simulink Control Design > Simulink Control Design > Simulink Control Design		

#### 2.1.1. Read ADC Channels—Outputting the input of an external ADC

Read ADC Channels: Select which ADC channel from the 3.3V and 6.6V analog inputs to be the output of this module, as shown in the figure below. Note that channels 1 and 2 correspond to p ins 14 and 15 of the 3.3V ADC, respectively.

Signal definition:

- 3.3V analog-to-digital conversion (int32)
- 3.3V Analog-to-Digital conversion (int32)
- 6.6V Analog-to-Digital (int32)

More information is available by clicking the "Help" button in the module dialog box.



#### 2.1.2. Serial—Serial Communication Module

Serial: As shown in the image below, serial communication block that allows reading and writ ing devices, specifying device name and read or write options. The specific usage can be checked by clicking the "Help" button of the module dialog box.

			Block Parameters: Serial	×
			SerialCOM	
			UART serial port System Object 源代码	t
_			 参数	
[			Device Name:	/dev/ttyPS0
			Read or Write to serial:	Writing
		Sorial LIAPT	Baud Rate:	115200
	Tx		Blocking	
•		/dev/ttyPS0	🗌 Enable Header	
			Enable Comm Protocol	
l			Simulate using:	Interpreted execution
		Serial		确定(0) 取消(C) 帮助(H) 应用(A)

# 2.2. Miscellaneous Utility Blocks-Other libraries

Simulink Library Browser							
💠 < mavlink 🗸 🗸 ka z 🔁 z 🤁 z 🔁 z							
ixhawk Target Blocks/Miscellaneous Utility Blocks							
Image Acquisition Toolbox Instrument Control Toolbox Model Predictive Control Toolbox > Neural Network Toolbox OPC Toolbox > Phased Array System Toolbox > Phased Array System Toolbox > Pixhawk Target Blocks MIC and Serial <u>Mice Claneous Utility Blocks</u> Sensors and Actuators uORB Read and Write > Powertrain Blockset Report Generator > RF Blockset > Robotic System Toolbox SimEvents > Simscape > Simulink Coder > Simulink Coder > Simulink Control Design Simulink Coder > Simulink Control Design							

## 2.2.1. binary\_logger—Data Logging Module

This module records the data as a double and stores it in the SD card, and if the cache is turne d on in memory, writes the data when en gets low or reaches the specified maximum number of rec ords. The first enable input signal during program execution must first trigger the high level and th en drop to the low level in order to record the data successfully. If the low level is not dropped bef ore the code finishes running, then the record file is considered to be open and inaccessible. In add ition, the log must be stored under the directory "/fs/microsd/". As shown in the following figure, y ou can set the path for storing logs and the maximum number of records. The specific usage can be clicked on the "Help" button of the module dialog box to view, and the specific routine experimen t can be seen in the file: <u>0.ApiExps\5.Log-Write-Read\Readme.pdf</u>

	Block Parameters: binary_logger         ×           Pixhawk Binary Logger (mask) (link)         Log a vector of data to a file on the SD card           If Cache in Memory is on data is written when en goes low or when max number of records specified is reached.         Otherrise Max Number of Records is used to indicate when to perform an order of the second s
▶ en /fs/microsd/log/mw	新知道       新知道         参数
<pre>&gt; data</pre>	<b>确定(0) 取消(C) 帮助(B)</b> 应用(A)

#### 2.2.2. ExamplePrintFcn—Print Function Example

As shown below, this module prints the signal data content to the PX4 Nuttx console terminal. This block should be considered an "example" block, and the print message can be constructed in any way the user sees fit. Coder.ceval() is a MATLABCoder function that evaluates printf() statem ents to pass two values. Note that there is a bug in Nuttx that sometimes floating-point numbers do not display correctly. Use warnx() instead of printf() as a workaround.



#### 2.2.3. ParamUpdate—Custom Storage Class Parameter Update Module

As shown in the image below, this module updates custom PX4 software parameters. Note wh en using: Assume that the csc is stored in your base workspace, other possible locations include a d ata dictionary or a model workspace. You will need to copy the appropriate csc into your base wor kspace before using this feature.





# 2.3. Sensors and Actuators—Sensor and Actuator Interface Library

## 2.3.1. Battery\_measure—Battery Data Module

This module allows you to obtain the battery health status by subscribing to messages posted by the uORB topic publisher battery\_status, so you need to ensure that the power module is plugge d into Pixhawk in order to obtain the correct data.

The signal definition:

- Voltage: (double) The battery voltage, in volts.
- Filtered Voltage: (single) Filtered voltage of the battery, in volts.
- Current: (single) Current of the battery, in amperes.
- mAH: (single) The amount of discharge in mAH.
- Timestamp: (int32) Timestamp of the measurement.

As shown in the image below, the module provides a setting box for the sample rate, as well a s selectable battery status options. More information can be viewed by clicking the "Help" button o f the dialog box.



#### 2.3.2. Input\_rc—Remote Control Input Module

This module allows the user to access the signal from the RC transmitter. Through this modul e, the output signal can be selected, including the value of multiple remote control channels, and so me other information. As shown in the following picture, these include:

- 1. Channel Selection Channel selection
  - a) uint16 data type, which represents the PWM(in use) value from the controller.
  - b) Measure the pulse width of each support channel.
- 2. Channel Count Number of channels
  - a) Uint32 Indicates the number of channels detected by the PX4 detector.
- 3. RC Failsafe Remote control signal failsafe
  - a) Boolean data type, indicating that RC Tx is sending FailSafe signal (if set correctly)
  - b) Displays the failsafe flag: true if Tx fails or if Tx is out of range, false otherwise.
  - c) Only the true state is reliable, as there are some (PPM) receivers on the market that go into failsafe without explicitly telling us.
- 4. RC Input Source Remote control signal input source
  - a) Enumerate the data type, indicating which source the RC input comes from.
  - b) Find valid values in the ENUM file:

RC\_INPUT\_SOURCE\_ENUM.m

RCINPUT_S	OURCE_	UNKNOWN	(0)
RCINPUT_S	OURCE_	PX4FMU_PPM	(1)
RCINPUT_S	OURCE	PX4IO_PPM	(2)
RCINPUT_S	OURCE	PX4IO_SPEKTRUM	(3)
RCINPUT_S	OURCE	PX4IO_SBUS	(4)

- 5. RSSI received signal strength indicator
  - Received Signal Strength Indicator (RSSI) : <0: undefined; 0: No signal; 255: full r eception.
- 6. RC Lost Connection Remote control signal lost connection
  - a) Boolean data type indicating RC receiver connection status.
  - b) True if no frame arrived in the expected time, false otherwise.
  - c) True usually means that the receiver is disconnected, but can also mean that the rad io link is lost on a "dumb" system.
  - d) If an RX with the failsafe option continues to transmit frames after the link is lost, it remains false.

More information can be viewed by clicking the "Help" button in the dialog box or by viewin g the official PDF document. See the file for specific routine experiment: <u>0.ApiExps\2.PSPOfficial</u>

#### Exps\Readme.pdf

		🚹 Block Paramet	ers: input_rc		×
		- PX4_Input_RC (	mask) (link) —		
		RC Input Block			
	Ch1	Receiver Input	s from the Pixhawk	k hardware	
		Sample Time			
		1/250			
		Channel Select	ion		
	Ch2	🖂 Channel 1	🗹 Channel 2	🕑 Channel 3	🔽 Channel 4
<u> </u>		Channel 5	Channel 6	Channel 7	Channel 8
1 / /		Channe19	Channel10	Channel11	Channel12
		Channel13	Channel14	Channel15	Channel16
	Ch D	Channel17		Channel18	
	Cha	- Optional Outpu	its		
		Channel Cou	nt	RSSI	
		🗌 RC Failsafe		🗌 RC Lost Con	nection
		🗌 RC Input So	urce		
	Ch4				
input ro					
input_rc			确定(0)	取消(C) ₹	幣助(H) 应用(A)

#### 2.3.3. PWM output—Motor PWM Module

Through this module, PWM signal can be sent to the output port of PX4IO to control the mot or rotation, you can select the PWM update rate and input channel.

In order for the flight control arm (enabled) to output from the software side, the ARM output input must be kept high (Boolean TRUE). Only then will the PWM value be sent to the PX4 hard ware port. This is usually a feature of the RC Tx combined with other flight modes programmed in the Simulink model by the user.

The block has 8 available ports (data type uint16) that can be selected selectively. These corre spond to the 8 PWM output ports on the px4fmu hardware.

The unit value of PWM is microsecond (usec), which corresponds to the pulse width (1500 is 1500usec or 1.5 ms).

The PWM update rate is set to 400Hz when the px4simulinkapp is started (or whatever the us er sets in the block dialog). The available PWM update rates are :50, 125, 250, 300, and 400Hz.

px4\_simulink\_app will also set an idle value of 900usec at startup and when the ARM port is set to low (Boolean FALSE) so that the ESC controller does not time out.

As shown below, for more information click the "Help" button of the dialog box to see.



# 2.3.4. RGB\_LED— LED Light

The mode and color of the LED light flashing can be controlled through this module. As show n below, the module receives two inputs, one is Mode, the other is Color, and these two inputs are enumerated data types. You can find a valid value in the MATLAB command window by entering the following command:

>>:enumeration ("RGBLED COLOR ENUM") Enumeration members of the 'RGBLEDCOLORENUM' class: COLOR OFF COLOR RED COLOR\_YELLOW COLOR PURPLE COLOR\_GREEN COLOR BLUE COLOR\_WHITE COLOR AMBER COLOR DIM RED COLOR\_DIM\_YELLOW COLOR DIM PURPLE COLOR DIM GREEN COLOR DIM BLUE COLOR DIM WHITE COLOR\_DIM\_AMBER Enumeration member of the 'RGBLED MODE ENUM' class: MODE OFF

MODE\_ON MODE\_BLINK\_SLOW MODE\_BLINK\_NORMAL MODE\_BLINK\_FAST MODE\_BREATHE MODE\_PATTERN

More information can be viewed by clicking on the "Help" button of the dialog box. See file f or specific routine experiment: <u>0.ApiExps\2.PSPOfficialExps\Readme.pdf</u>



#### 2.3.5. sensor combined—Sensor Fusion Module

It is through this module that the sensor data available in Pixhawk can be obtained, which can then be used to control the design of the model. The data available includes magnetometers, accel erometers, gyroscopes, barometers, and time stamps.

Signal definition:

- Magnetometer(x,y,z):(single) Magnetometer(x,y,z), magnetic field under NED, Gauss un it.
- Accelerometer(x,y,z): (single) accelerometer (x,y,z), the frame of acceleration under NE D, in m/s^2.
- Gyroscope(x,y,z): (single) gyroscope (p,q,r), angular velocity in radians per second.
- Barometer(Altitude): (single) barometer (altitude), air pressure with temperature compen sated (mbar).
- uORB and RunTime(timestamp):(double) Time stamp in microseconds since the gyrosc ope was activated.

The sensor\_combined block needs to run the px4io service on the PX4 hardware in order to g et a valid signal value.

As shown below, more information can be viewed by clicking the "Help" button of the dialog box. See file for specific routine experiment: <u>0.ApiExps\11.SenorDataGet\Readme.pdf</u>



#### 2.3.6. Speaker\_Tune—Buzzer Module

Through this module, you can control the buzzer to emit a specific tone at a specific event. As shown below, this module accepts 3 input signals.

- Tune ID: This is the enumerated data type of predefined "tunes" that can be played. Enu meration members of class 'PX4 TUNE ENUM' :
  STOP TUNE
  STARTUP TUNE
  ERROR TUNE
  NOTIFY\_POSITIVE\_TUNE
  NOTIFY\_NEUTRAL\_TUNE
  NOTIFY\_NEGATIVE\_TUNE
  ARMING\_WARNING\_TUNE
  BATTERY\_WARNING\_FAST\_TUNE
  GPS\_WARNING\_TUNE
  ARMING\_FAILURE\_TUNE
  PARACHUTE RELEASE TUNE
- IsTuneOverride: Used to define a custom tune to play. The user can use a constant block to define the string to be played.
- Trigger: This is a trigger signal that indicates when a predefined tune (if the trigger value changes to 1) or a custom tune (if the trigger value changes to 2) will be played. A chan ge in the tune will only be triggered by a change in that value.

More information is available by clicking on the "Help" button of the dialog box.



#### 2.3.7. vehicle attitude—Attitude Data Module

This module provides filtered attitude data (Euler Angle and quaternion) and provides access to a running service that calculates the attitude of the UAV. A uORB theme (vehicle attitude(Attitu de Measurement)) publisher must be run in order for the block to provide valid signal values.

One of these must be run on px4fmu in order for the block to return a valid value. For exampl

e:

px4fmu-v2 ekf2: ekf Extended Kalman filter for attitude estimation

px4fmu-v2 default: SO(3)- Attitude estimation using accelerometers, gyroscopes and magneto meters

As shown in the image below, the module provides a setting box for the sampling rate, as wel l as selectable attitude options.

Signal definition:

- Roll rate:(single) Roll rate in degrees per second or radians per second (NED).
- Pitch rate: (single) Pitch rate in degrees per second or radians per second (NED).
- Yaw rate: (single) yaw rate in degrees/second or radians/second (NED).
- Quaternion(NED): (single) Optional according to uORB publisher (NED).
- uORB Ts: Used for efficient data exchange and time synchronization.

For more information, click the "Help" button in the dialog box.



#### 2.3.8. vehicle gps— GPS Data Module

The Pixhawk GPS data can be accessed through this module, which is achieved by subscribin g to the uORB hashtag "vehicle\_gps", so you will need to ensure that the GPS module is plugged i nto Pixhawk to get the correct data during the actual run. As shown in the image below, the modul e provides a setting box for the sample rate, as well as selectable attitude options. The meaning of each option is as follows:

- Latitude: (int32) Global coordinates are given at 1e-7 degrees.
- Longitude: (int32) Longitude is given in 1e-7 degrees.
- Altitude: (int32) is 1e-3 m (mm) above MSL (mean sea level).
- GPS TS: (double) Timestamp (microseconds in GPS format), this is the timestamp from the GPS module.
- Velocity: (single)GPS ground speed, in meters per second.
- GPS Fix Type: (uint8)0-1=NO fix,2=2D fix,3=3D fix.
- Num Satellites: (uint8) Number of satellites used to calculate.

More information is available by clicking the "Help" button in the dialog box.

Datitude	Block Parameters: vehicle_gps           PX4_Vehicle_GPS (mask) (link)	×
Longitude	Return Values from the GPS topic running on the Pixhawk. Sample Time	
	1/250	
Aittude	Output Selection	
	✓ Altitude	
GPS TS	✓ GPS Timestamp	
	Velocity	
Velocity	✓ Fix Type	
	✓ Number of Satellites	
GPS Fix Type		
vehicle_gps	确定(0) 取消(C) 帮助(H) 应用	(A)

# 2.4. uORB Read and Write— uORB Message Read and Write Libra

#### ry



#### 2.4.1. uORB Read Async—Retrieve Data Related to uORB Topic

As shown in the image below, the block will get data related to the uORB topic that drives the asynchronous subsystem where the block is located. The topic name will be determined automatic ally based on the callback, and the bus object will need to be specified for the topic. More informat ion can be found by clicking the "Help" button of the dialog box.



# 2.4.2. uORB Read Function-Call Trigger— uORB Message Read Callback Trigg er Module

This module provides two functions, the first of which is to subscribe to messages from a uO RB topic. The second is to subscribe to message data on a topic by triggering a function call signal for asynchronous events.



# uORB Read Function-Call Trigger

Block Parameters: uORB Read Function-Call Trigger	×
uORB Read and Function-Call Trigger	
This block operates in two modes:	
Mode1: Reading uORB Data	
In this mode, the block will fetch data using a uORB_CHECK() call and write the data to a struct. This struct is accessed as a bus object. The buttons below allow one to create bus objects for any arbitrary uOI message provided it is specified in a .msg format. Header files are added to the generated code from: C: \px4\Firmware\src\modules\uORB\topics	RB
Mode2: Asynchronous Tasking	
In this mode, the block will spawn a thread which will be data-driven. When a uORB message of the specified topic arrives, the thread will advance in execution and run the contents of the affiliated function-call subsystem	
uORB Topic differential pressure	:
Open .msg file Select uORB Topic msg Create Bus Object	
Enter Bus Object 🗛 Bus: differential_pressure_SL 🗸 🗸 💛	
uORB Interval (ms) 1	
Function-Call Options	
✓ Function-Call Trigger	
TaskName 'Task1'	
Idaki	
Polling Timeout (ms) (-1 = indefinite wait, 0 = return immediately)	
确定(0) 取消(C) 帮助(H) 应用(	()

As shown in the figure above, the steps of using the first function are:

• Choose a defined topic

Click the button "Select uORB Topic msg" to open the list of topics for selection, only to pics that are not C++ objects are supported.

• Create the bus (bus) object

Simulink's Bus object is used to receive uORB messages, click the button "Create Bus O nject" and Simulink will find the corresponding message file from the.msg folder and map it t o the MATLAB workspace to generate the bus object.

• Set the uORB read interval

In non-asynchronous mode, you need to set the query frequency in milliseconds. For som e topics, the maximum data update rate is set. Do not set the frequency to exceed this maximu m.

The steps to use the second function are as follows:

Select the function call trigger

Function-Call Options Function-Call Trigger	
TaskName 'Task1'	
Polling Timeout (ms) (-1 = indefinite wait, $0$ = return immediately) 100	

• Set the query timeout and task name

When asynchronous is selected, the sampling time setting box disappears, and you need to set the query timeout parameter and task name. The asynchronous function spawns a new t hread that runs the code associated with the function trigger signal and waits for new data on t he topic by querying it. At this point, another module is needed to read the data on the topic, n amely the "Read uORB function Trigger Data Module", as shown in the figure below.



#### 2.4.3. uORB Write— uORB Message Data Publishing Interface Module

This interface allows users to publish specified values or structs to uORB topics. This module allows users to publish messages to a uORB topic. Topics must be properly defined, and some def ined topics are placed in the C:\PX4PSP\Firmware\msg directory, which automatically contains th e topic definition file in the generated code.

As shown in the following figure, you can enter the topic name, click the button "Open. msg f ile" to Open the corresponding message content, click the button "open. msg folder" to Open the to pic list, and set the input port name and data type to correspond to the topic message.



Block Parameters: uORB Write					×
-S-Function (mask) (link)					
uORB Write Output Block					
Publishes to a user provided name	ed tropic structure.				
uORB Topic 'actuator_outputs'	!	Open.msg file		Open .msg folder	
Number of Outputs 2					:
uORB Parameter Names and Data Type					
✓ Input Struct param :					
'output'	: single		>> Numbe	er of Flements 32	
output			//		
✓ Input Struct param :					
'noutputs'	i uint32	<u>~</u> :	>> Numbe	er of Elements 1	
□ Input Struct param :					
□ Input Struct param :					
□ Input Struct param :					
Sample Time (-1 inherited) _1					
			确定(0)	取消(C) 赛肋(H)	应田(A)
			MUNE (0)		1.04.7 (11)
ActuatorOutputs.msg × +	# +;	me since system	start (micro	seconds)	
2 uint8 NUM ACTUATOR OL	JTPUTS = 16	me since system	start (mittro	查看消息	内容
		- 4 # fon conity	checking		
3 uint8 NUM_ACTUATOR_OL	JTPUT_GROUPS	= 4 # TOP Sanity	checking		
<pre>uint8 NUM_ACTUATOR_OU uint32 noutputs float32[16] output</pre>	JTPUT_GROUPS # valid # ou	outputs	tural output	units	
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6</pre>	JTPUT_GROUPS # valid # ou	outputs tput data, in na	tural output	units	
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 6 7 # TOPTCO Functional Statement Statemen</pre>	JTPUT_GROUPS # valid # ou im is used for S	outputs tput data, in na ITL, HITL & SIH	tural output (with an out	units put range of [-1, 1])	)
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9</pre>	JTPUT_GROUPS # valid # ou im is used for S :puts actuator_o	outputs tput data, in na ITL, HITL & SIH utputs_sim actua	(with an out tor_outputs_	units put range of [-1, 1]) debug	)
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9</pre>	JTPUT_GROUPS # valid # ou im is used for S :puts actuator_o	outputs tput data, in na ITL, HITL & SIH utputs_sim actua	tural output (with an out tor_outputs_	units put range of [-1, 1]) debug	)
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 1 Block Parameters: uORB Write Schumetion (mark) (link)</pre>	JTPUT_GROUPS # valid # ou im is used for S :puts actuator_o	outputs tput data, in na ITL, HITL & SIH utputs_sim actua	tural output (with an out tor_outputs_	units put range of [-1, 1]) debug	)
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 1 Block Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block</pre>	JTPUT_GROUPS # valid # ou im is used for S :puts actuator_o	outputs tput data, in na ITL, HITL & SIH utputs_sim actua	(with an out (with an out (tor_outputs_	units put range of [-1, 1]) debug	)
<pre>3 uint8 NUM_ACTUATOR_OU 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 Pa Block Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block Publishes to a user provided named to </pre>	JTPUT_GROUPS # valid # ou im is used for S :puts actuator_o	outputs tput data, in na ITL, HITL & SIH utputs_sim actua	(with an out tor_outputs_	units put range of [-1, 1]) debug	) ×
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 Pa Block Parameters: uORB Write S-Function (mask) (link) u0RB Write Output Block Publishes to a user provided named t u0RB Topic 'actuator_outputs'</pre>	JTPUT_GROUPS # valid # ou im is used for S cputs actuator_o	Outputs tput data, in na ITL, HITL & SIH utputs_sim actua	(with an out (with an out tor_outputs_	units put range of [-1, 1]) debug Open.msg folder	)
<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 Pa Block Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block Publishes to a user provided named t uORB Topic 'actuator_outputs' Number of Outputs 2</pre>	JTPUT_GROUPS # valid # ou im is used for S puts actuator_o	Open.msg file	(with an out (with an out itor_outputs_	units put range of [-1, 1]) debug Open .msg folder	) ×
<pre>3 uint8 NUM_ACTUATOR_OU 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 9 1 1 2 Block Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block Publishes to a user provided named t uORB Topic 'actuator_outputs' Number of Outputs 2 uORB Parameter Names and Data Type</pre>	JTPUT_GROUPS # valid # ou im is used for S :puts actuator_o	Open .msg file	(with an out (with an out tor_outputs_	units put range of [-1, 1]) debug Open .msg folder	)
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<pre>3 uint8 NUM_ACTUATOR_OL 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 Page Block Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block Publishes to a user provided named t uORB Topic <u>'actuator_outputs'</u> Number of Outputs 2 uORB Parameter Names and Data Type &gt; 此地職論 &gt; 系统 (C.) &gt; PX4PSP &gt; Firm </pre>	JTPUT_GROUPS # valid # ou im is used for S tputs actuator_o	Open.msg file	vitural output (with an out itor_outputs_	e units put range of [-1, 1]) debug Open .msg folder	۲ ۱ ۲
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<pre>3 uint8 NUM_ACTUATOR_OU 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1</pre>	JTPUT_GROUPS	Performantly     outputs     tput data, in na     ITL, HITL & SIH     utputs_sim actua      Open .msg file      Open .m	c/v (with an out tor_outputs_	e units put range of [-1, 1]) debug Open .msg folder 了在msg 中媲素 軍、 打开话题列表	× × ۹
<pre>3 uint8 NUM_ACTUATOR_OU 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 Polick Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block Publishes to a user provided named to uORB Topic 'actuator_outputs' Number of Outputs 2 uORB Parameter Names and Data Type •</pre>	JTPUT_GROUPS	Performantly     outputs     tput data, in na     ITL, HITL & SIH     utputs_sim actua      Open .msg file      Open .msg file      使型     反utlook.File.msg     Outlook.File.msg	<pre>clicking itural output (with an out itor_outputs_</pre>	e units put range of [-1, 1]) debug Open .msg folder C 在msg 中搜索 軍 、 打开话题列表	× × ۹
<pre>3 uint8 NUM_ACTUATOR_OU 4 uint32 noutputs 5 float32[16] output 6 # actuator_outputs_si 8 # TOPICS actuator_out 9 # OPICS actuator_out 9 # Block Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block Publishes to a user provided named to uORB Topic <u>'actuator_outputs'</u> Number of Outputs 2 uORB Parameter Names and Data Type •</pre>	JJTPUT_GROUPS # valid # ou im is used for S tputs actuator_o im is structure. [] wware > msg 修改日期 2024/3/22 15:13 2024/3/22 15:13	中 W FOF Samily         outputs         tput data, in na         ITL, HITL & SIH         utputs_sim actua         Open .msg file         Øpen .msg file         Øpen .msg file         Øpen .msg file         Øutlock.File.msg         Outlock.File.msg	<pre>checking itural output (with an out itor_outputs_</pre>	e units put range of [-1, 1]) debug Open .msg folder C 在 msg 中继案 軍 マ 打开话题列表	× :: ۹
<pre>3 uint8 NUM_ACTUATOR_OU 4 uint32 noutputs 5 float32[16] output 6 7 # actuator_outputs_si 8 # TOPICS actuator_out 9 Block Parameters: uORB Write S-Function (mask) (link) uORB Write Output Block Publishes to a user provided named 1 uORB Topic <u>'actuator_outputs'</u> Number of Outputs 2 uORB Parameter Names and Data Type • 此电脑 &gt; 系统 (Ci) &gt; PX4PSP &gt; Firm ※ Actuator Names and Data Type •</pre>	JTPUT_GROUPS	4 # FOF Samily         outputs         tput data, in na         ITL, HITL & SIH         utputs_sim actua         Open .msg file         Open .msg file         Quenter file	<pre>clicking itural output (with an out itor_outputs_</pre>	z units put range of [-1, 1]) debug Open .msg folder C 在 msg 中强素 重 マ 打开话题列表	× :: ۹

# 2.4.4. uORB Write Advanced— Advanced Module for uORB Message Data Pub lishing Interface

This interface allows users to have more flexible control over the data they publish. Using the uORB Write Advanced interface in Simulink, a more complex and precise way of publishing mess ages can be achieved. You can select the message file and a message ID to write to. In addition, yo u can also set the priority of the message, queue size and other advanced options.



As shown in the following picture, you can see the name of the current topic, click the button "Open. msg file" to Open the corresponding message content, and click the button "Select. msg fil e" to open the topic list, and you can set the input port name and data type to correspond to the topic c message.

					- 0 X
uORB Me	essage: actuator_outputs	Apply	Select .msg file	Open .msg file	
	Struct Field	Dimensions DataType Enable			
	2 noutputs	1 uint32 ~			
	3 output	16 single V			
			C Enable	Disable All	
			_		
		Advertisement Queue (0	= no queue) :	0	
			Sample Time:	.4	
			Okay	Cancel	
ActuatorOutpu	uts.msg × +				
uint64	timestamp	# time since sy	stem start (mi	croseconds)	本手巡自市。
	NUM_ACTUATOR_OUTPUTS	= 16			旦有/月尽内1
urnro			anity checking		
uint8	NUM_ACTUATOR_OUTPUT_C	GROUPS = 4 # tor s	anity checking		
uint8 uint32	NUM_ACTUATOR_OUTPUT_C	GROUPS = 4 # for s # valid outputs	anity checking		
uint8 uint32 float3	NUM_ACTUATOR_OUTPUT_0 noutputs 2[16] output	GROUPS = 4 # for s # valid outputs # output data,	in natural out	put units	
uint8 uint32 float3	NUM_ACTUATOR_OUTPUT_0 noutputs 2[16] output	GROUPS = 4 # for s # valid outputs # output data,	in natural out	put units	
uint8 uint32 float3 # actu	NUM_ACTUATOR_OUTPUT_( noutputs 2[16] output ator outputs sim is u	GROUPS = 4 # for s # valid outputs # output data, used for SITL. HITL &	in natural out	put units putput range of	[-1, 1])
uint8 uint32 float3 # actu # TOPT	NUM_ACTUATOR_OUTPUT_( noutputs 2[16] output ator_outputs_sim is u CS actuator outputs a	<pre>sROUPS = 4 # for s     # valid outputs</pre>	in natural out SIH (with an actuator output	put units output range of ts debug	[-1, 1])

ORB Write Block						-
uORB Message:	actuator_outputs		Apply	Select .msg file	Open .msg file	
1 timot	Struct Field	Dimensions DataType	Enable			
(电脑 > 系统(C:) > PX4PSP >	Firmware > msg			~	C 在 msg 中搜索	
					≣ ▼	
名称 ^	修改日期	类型	大小			
ActionRequest.msg	2024/3/22 15:13	Outlook.File.msg	1 KB		打开话题列表	
ActuatorArmed.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			
ActuatorControlsStatus.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			
ActuatorMotors.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			
ActuatorOutputs.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			
ActuatorServos.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			
ActuatorServosTrim.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			
ActuatorTest.msg	2024/3/22 15:14	Outlook.File.msg	1 KB			
AdcReport.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			
Airspeed.msg	2024/3/22 15:13	Outlook.File.msg	1 KB			

# 2.4.5. uORB Write Advanced\_dai— Advanced Module for uORB Message Data

# **Publishing Interface**

Compared with uORB Write Advanced, uORB Write Advanced\_dai adds the function of cust omizing uORB MsgID.

<b>&gt;</b> output	To uOF	uORE pic: ac	3 Write_dai tuator_outputs te Advanced_dai							
RB Write Block									-	
uORB Messa	age:		actuator_outputs		A	ply	Select .msg file	Ope	n .msg file	
uORB Ms	gID:		actuator_outputs_0							
			Struct Field	Dimensions	DataType	Enable			]	
	1	timesta	mp	1	double ∨					
	3	output	3	16	single ~					
							Enat	le/Disable All	]	
					Adverti	sement Queue	() = no queue) :	le/Disable All		
					Advertis	sement Queue	() = no queue) : Sample Time:	le/Disable All 0 -1		

See the file for the specific routine experiment:

0.ApiExps\5.Log-Write-Read\Readme.pdf

0.ApiExps\6.uORB-Read-Write\Readme.pdf

# 3. MATLAB Command Line Interface

The RflySim platform also supports running relevant commands through the MATLAB comm and line window, including:

# 3.1. PX4Upload

You can upload the PX4 firmware to the flight controller with a single click. The firmware up loaded at this time is located at: \*\PX4PSP\Firmware\build\[build command]\[build command].px 4 (For example, [build command] could be px4\_fmu-v6x\_default).

PX4Upload

After executing the above command, a black window will pop up, prompting the user to plug and unplug the flight controller, and displaying the upload progress bar.

## **3.2. PX4CMD**

To switch to the compilation environment for the Pixhawk 6C flight controller, you can perfor m the following steps to replace the firmware compilation options:

```
PX4CMD('px4_fmu-v6c_default')
或
PX4CMD 'px4_fmu-v6c_default'
```

# 3.3. PX4Build

Firmware compilation is possible.

## 3.4. PX4AppName

Rename the PX4 software's APP to support multiple automatic code generation programs. For detailed usage, please refer to:

```
PX4AppName('rfly_simulink_app')
```

**%** 或

PX4AppName 'rfly\_simulink\_app'

Related examples can be found at: <u>2.AdvExps\e0\_AdvApiExps\1.CusMaskPX4Code\Re</u> adme.pdf, <u>2.AdvExps\e0\_AdvApiExps\2.RenamePX4App\Readme.pdf</u>

# 3.5. PX4AppLoad

Load the renamed PX4 software's App to import previously developed App programs. The us age is as follows:

```
PX4AppLoad('C:\PX4PSP\rfly_simulink_app')
```

PX4AppLoad 'C:\PX4PSP\rfly\_simulink\_app'

Related examples can be found at: <u>2.AdvExps\e0\_AdvApiExps\1.CusMaskPX4Code\Re</u> <u>adme.pdf</u>, <u>2.AdvExps\e0\_AdvApiExps\3.LoadPX4App\Readme.pdf</u>

#### 3.6. PX4ModiFile

To replace parts of the code in PX4 software through Excel, follow these steps:

PX4ModiFile('C;\Users\dream\Desktop\自定义屏献 UORB 消息的例子 px4Block.xlsx')

Related examples can be found at: <u>2.AdvExps\e0\_AdvApiExps\1.CusMaskPX4Code\Re</u>

adme.pdf, 2.AdvExps\e0 AdvApiExps\2.RenamePX4App\Readme.pdf

### **3.7.** PX4Official

By executing the command, you can directly generate official firmware (without output mask ing), which can be used to restore the flight controller for HITL external control or to repair proble matic flight controllers. The usage is as follows:

```
PX40fficial
执行完上述命令后,再输入下面指令,可以将官方固件上传到飞控中:
PX4Upload
```

#### 3.8. PX4SitlSet

To enable the current automatically generated controller, px4\_simulink\_app, to support SITL simulation, follow these steps: In the Simulink program, click "Build" to generate the hardware-in-the-loop (.px4) file. Run PX4SitlSet directly after generating the .px4 file.Execute SITLRun (for st andard quadcopter) or other SITL simulation scripts driven by the DLL model to simulate the hard ware-software interaction of the automatically generated algorithm.

Command format:

PX4SitlSet

Related examples can be found at: <u>0.ApiExps\14.SITLVeriGenCodeFirm\Readme.pdf</u>

#### 3.9. PX4SitlRec

In the SITL simulation code, remove the automatically generated controller, px4\_simulink\_ap p, to revert to the normal software-in-the-loop simulation mode, which supports QGC control and Offboard external control. Note: After testing the Simulink controller by running PX4SitlSet, if yo u want to run the platform's official vision or external control routines again, please use PX4SitlRe c to restore the environment first.

Command format:

PX4SitlRec

Related examples can be found at: <u>0.ApiExps\14.SITLVeriGenCodeFirm\Readme.pdf</u>

# 4. Automatically generated external communication interfa

#### ce

When running the one-click installation script for RflySim, the platform modifies the source c ode in the Firmware directory by adding four uORB messages. These messages are registered in \*\

PX4PSP\Firmware\msg\CMakeLists.txt\*. Detailed information and usage guidelines are provided below:

# 4.1. rfly\_ctrl.msg

Format of messages transmitted from external sources into PX4 via UDP or MAVLink protoc

```
ol:
```

uint64 timestamp	<pre># time since system start (microseconds)</pre>
uint32 flags	# control flag
uint8 modes	# mode flag
float32[16] controls	# 16D control signals
External data transmission	from Simulink to internal uORB messages in PX4:

Transmitter: Refer to the module "SendToPX4UorbRflyCtrl" in the example "0.ApiExps\9.PX

<u>4CtrlExternalTune\Readme.pdf</u>". This module can send messages through the 30100 series port to CopterSim, and then forward them to the internal uORB message "rfly\_ctrl" in PX4 through the M AVLink protocol.



Receiver: When designing low-level controllers in Simulink, simply subscribe to the "rfly\_ctr l" message to receive the data.



From port 30100 to receive external control signals

Experimental Principle: After receiving the "rfly\_ctrl" message, PX4ExtMsgReceiver.slx sim ulates the first five elements of "controls" as inputs from the remote controller, which are then fed into the attitude controller of the previous example to perform attitude control. Therefore, PX4Ext MsgSender.slx can simulate sending inputs from the remote controller to test the attitude algorithm.

Experimental Procedure: PX4ExtMsgReceiver.slx is burned into the flight controller through automatic code generation and launched for hardware-in-the-loop simulation. Then, PX4ExtMsgS ender.slx is opened in Simulink. By adjusting the sliders on the left side of the "SendToPX4UorbR flyCtrl" block, various channel data from the remote controller can be simulated to control the dro ne.

**Communication Principle from Simulink to CopterSim:** Analyzing the "SendToPX4Uorb RflyCtrl" module reveals that this Simulink module sends the following structure via UDP to the 1 27.0.0.1:30100 port on the local machine.

```
struct PX4UorbRflyCtrl {
  int checksum;
  int CopterID;
  uint32_t modes;
  uint32_t flags;
  float data[16];
}
```

Note: The checksum here must be set to 1234567896 in order to pass the verification by Copt erSim.



→ CH1 → CH2 → CH3 udp → CH4	UDP Send UDP30100	
Ch5 SendToPX4UorbRflyCtrl	Block Parameters: UDP30100 UDP Send (mask) (link)	×
	Send a UDP packet to a network address identified by the remote IP address and remote IP port parameters.	
	Parameters Remote IP address ('255.255.255.255' for broadcast):	
	'127.0.0.1' Remote IP port:	
g rfly_ctrl through port 30100	30100 Local IP port source: Automatically determine	•
	OK Cancel Help Appl	y

**Communication Principle from CopterSim to PX4:** After receiving the PX4UorbRflyCtrl messa ge, CopterSim will further relay it as the "hil\_actuator\_controls" MAVLink message (https://mavli nk.io/en/messages/common.html#HIL\_ACTUATOR\_CONTROLS), and then forward it to the PX 4 flight controller. The definition of this message is as shown in the following figure.

#### HIL\_ACTUATOR\_CONTROLS (#93)

[Message] Sent from autopilot to simulation. Hardware in the loop control outputs (replacement for HIL\_CONTROLS)

Field Name	Туре	Units	Values	Description
time_usec	uint64_t	us		Timestamp (UNIX Epoch time or time since system boot). The receiving end can infer timestamp format (since 1.1.1970 or since system boot) by checking for the magnitude of the number.
controls	float[16]			Control outputs -1 1. Channel assignment depends on the simulated hardware.
mode	uint8_t		MAV_MODE_FLAG	System mode. Includes arming state.
flags	uint64_t			Flags as bitfield, 1: indicate simulation using lockstep.

Note: In reality, during hardware-in-the-loop simulation, this message is used by PX4 to trans mit motor control commands to CopterSim. Here, it is borrowed to transmit data back to PX4.

The RflySim platform has modified the source code at C:\PX4PSP\Firmware\src\modules\ma vlink\mavlink\_receiver.cpp to add support for the hil\_actuator\_controls message.

Internal parsing of the rfly\_ctrl message by PX4: As seen in the source code below, both rf ly\_ctrl and rfly\_ext borrow the hil\_actuator\_controls message for data transmission. When the mo de is 123, data is sent to the Simulink controller via rfly\_ext, while in other cases, it is sent to the c ontroller via rfly\_ctrl.

C: > PX4	PSP > Firmware > src > modules > mavlink > C+ mavlink_receiver.cpp
109	MavlinkReceiver::handle_message(mavlink_message_t *msg)
110	{
111	<pre>switch (msg-&gt;msgid) {</pre>
112	<pre>case MAVLINK_MSG_ID_HIL_ACTUATOR_CONTROLS:{</pre>
113	<pre>mavlink_hil_actuator_controls_t hil_actuator_control;</pre>
114	<pre>mavlink_msg_hil_actuator_controls_decode(msg, &amp;hil_actuator_control);</pre>
115	<pre>if(hil_actuator_control.mode==123){</pre>
116	<pre>rfly_ext_s re{};</pre>
117	<pre>re.timestamp = hrt_absolute_time();</pre>
118	re.modes = 1;
119	<pre>for(int i=0;i&lt;16;i++){</pre>
120	<pre>re.controls[i]=hil_actuator_control.controls[i];</pre>
121	}
122	<pre>_rfly_ext_publish(re);</pre>
123	}else{
124	<pre>rfly_ctrl_s rc{};</pre>
125	<pre>rc.timestamp = hrt_absolute_time();</pre>
126	<pre>rc.modes = hil_actuator_control.mode;</pre>
127	<pre>rc.flags = hil_actuator_control.flags;</pre>
128	<pre>for(int i=0;i&lt;16;i++){</pre>
129	<pre>rc.controls[i]=hil_actuator_control.controls[i];</pre>
130	}
131	<pre>_rfly_ctrl_pub.publish(rc);</pre>
132	}
133	break;
134	}

**Python Interface Usage:** According to the forwarding principle of CopterSim, we can send t he PX4UorbRflyCtrl structure to port 30100 or directly send MAVLink messages to the PX4 flight controller. There are two methods to input the rfly\_ctrl message.

First Method: Utilize the sendPX4UorbRflyCtrl function from the PX4MavCtrlV4.py interfac e for data transmission.

~~	
23	ctrls=[1500,1500,1900,1500,1900,1100,1
24	<pre>mav.sendPX4UorbRflyCtrl(ctrls)</pre>
25	# 发送SendToPX4UorbRflyCtrl数据,在PX4内
26	# 本消息主要设置油门通道(3通道)置于最高,
27	

The sendPX4UorbRflyCtrl function internally sends the PX4UorbRflyCtrl structure to the 30 100 series port, which is then forwarded to the PX4 flight controller.

1258	# }2i2I16f
1259	<pre>def sendPX4UorbRflyCtrl(self,data=[0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0,0],modes=1,flags=1):</pre>
1260	checksum=1234567896
1261	<pre>buf = struct.pack("2i2I16f",checksum,self.CopterID,modes,flags,*data)</pre>
1262	<pre>self.udp_socket.sendto(buf, (self.ip, self.port+10000))</pre>

Since we know that both rfly\_ctrl and rfly\_ext are Mavlink messages originating from hil\_act uator\_controls to transmit data, we can also directly use Python to send Mavlink messages for data transmission.

```
22
 23
      24
      mav.SendHILCtrlMsg(ctrls)
      # 发送SendToPX4UorbRflyCtrl数据,在PX4内部产生rfly_ctrl的uORB消息
 25
 26
      # 本消息主要设置油门通道(3通道)置于最高,飞机起飞。同时5通道置于最高,控制器解锁。
2213
         # send hil_actuator_controls message to Pixhawk (for rfly_ctrl uORB message)
2214
         def SendHILCtrlMsg(self,ctrls):
            """ Send hil_actuator_controls command to PX4, which will be transferred to uORB message rfly_ctrl
2215
            https://mavlink.io/en/messages/common.html#HIL_ACTUATOR_CONTROLS
2216
2217
2218
            time_boot_ms = int((time.time()-self.startTime)*1000)
            2219
2220
            for i in range(len(ctrls)):
2221
               if i<len(controls):</pre>
                 controls[i]=ctrls[i]
2222
            if self.isCom or self.isRealFly:
2223
2224
               self.the_connection.mav.hil_actuator_controls_send(time_boot_ms,controls,1,1)
2225
            else:
2226
               buf = self.mav0.hil_actuator_controls_encode(time_boot_ms,controls,1,1).pack(self.mav0)
2227
               self.udp_socket.sendto(buf, (self.ip, self.port))
2228
            #print("Msg Send.")
2229
```

#### 4.2. rfly\_ext.msg

Transferring directly from the DLL model into the PX4 internals, the message definition form at is:

uint64 timestamp	<pre># time since system start (microseconds)</pre>
uint8 modes	# mode flag
float32[16] controls	# 16D control signals

1) During DLL model development, the output port ExtToUE4PX4 is introduced. The first 1 6 dimensions of data from this interface are directly sent to UE4 for actuator control execution, wh ile the subsequent 16 dimensions are directly sent to PX4. These can be received in the low-level c ontroller via the rfly ext message.



2) When loading the DLL model in CopterSim, the 32-dimensional data from ExtToUE4PX4 is split into a 16-dimensional ExtToUE4 structure and a 16-dimensional ExtToPX4 structure. The l atter is then passed into the PX4 flight controller via the MAVLink message hil\_actuator\_controls. It's important to note that a passphrase "mode=123" is forcibly set during this transmission. Theref ore, as mentioned earlier, mavlink\_receiver.cpp will parse it as rfly\_ext before forwarding it.

3) In the Simulink low-level controller, subscribing to "rfly\_ext" allows for receiving data fro m the DLL model. This interface can be used to transmit sensor data that CopterSim currently does

not support to PX4, thus accelerating the debugging speed of hardware-in-the-loop simulation env ironments.



4) Testing Method: PX4ExtMsgReceiver.slx automatically generates code and burns it into t he flight controller. Exp2\_MaxModelTemp.slx generates the DLL model and enables hardware-in-the-loop simulation. In the DLL model, numbers 17 through 32 are sent to rlfy\_ext, and then the lo w-level controller forwards them as rfly\_px4 UORB messages. Therefore, the method described la ter can be used to view the data in QGroundControl (QGC) to verify if the messages from the DLL model are correctly transmitted.

# 4.3. rfly\_px4.msg

To transmit PX4 internal data externally through UDP or MAVLink protocol, the message def inition format is as follows:

uint64 timestamp	<pre># time since system start (microseconds)</pre>
float32[8] control	# 8D control signals

 Publish the rfly\_px4 message in the low-level controller, then subscribe to the data throug h port 40100 in Simulink. Refer to the example in the UDP\_SIL\_State\_Receiver2 module located i n <u>0.ApiExps\9.PX4CtrlExternalTune\Readme.pdf</u>.



Get desired signals from PX4 through rfly\_px4 uORB API with port 40101

2) The RflySim platform has modified the "Firmware\src\modules\mavlink\streams\ACTUA TOR\_CONTROL\_TARGET.hpp" file to subscribe to rfly\_px4 and forward it as ACTUATOR\_CO NTROL\_TARGET MAVLink messages, which are then received by CopterSim.



Please refer to the detailed definition of the message: <u>https://mavlink.io/en/messages/commo</u> <u>n.html#ACTUATOR\_CONTROL\_TARGET</u>

#### ACTUATOR\_CONTROL\_TARGET (#140)

[Message] Set the vehicle attitude and body angular rates.

Field Name	Туре	Units	Description
time_usec	uint64_t	us	Timestamp (UNIX Epoch time or time since system boot). The receiving end can infer timestamp format (since 1.1.1970 or since system boot) by checking for the magnitude of the number.
group_mlx	uint8_t		Actuator group. The "_mlx" indicates this is a multi-instance message and a MAVLink parser should use this field to difference between instances.
controls	float[8]		Actuator controls. Normed to -1+1 where 0 is neutral position. Throttle for single rotation direction motors is 01, negative range for reverse direction. Standard mapping for attitude controls (group 0): (index 0-7): roll, pitch, yaw, throttle, flaps, spoilers, airbrakes, landing gear. Load a pass-through mixer to repurpose them as generic outputs.

3) After receiving the ACTUATOR CONTROL TARGET message, CopterSim will check i f group mlx equals the code 123. If it does, it will forward the following structure to the 40100 ser ies port.

```
struct PX4ExtMsg {
   int checksum; //1234567898
   int CopterID;
    double runnedTime; //Current stamp (s)
float controls[8];
}
```

Note: CopterSim uses odd-numbered ports for outgoing communication and even-numbered p orts for incoming communication. Therefore, for the first aircraft, when publishing the PX4ExtMs g message to the corresponding 40100 series port, it is actually using port 40101.



Get desired signals from PX4 through rfly\_px4 uORB API with port 40101

After listening to port 40101 through the UDP module, the data is parsed and outputted.



In the parsing function, checks are performed on the length and checksum flags to prevent rea ding incorrect data. Additionally, if the correct data is not received, the aforementioned data will be retained.

1	编	輯器	록 - Block: PX4ExtMsgSender/UDP_SIL_State_Receiver2/
	ι	JDP_	SIL_State_Receiver2/VehicleDataPerse 🗙 🕂
1		E	function [y, data0k] = fcn(Data, Length)
2	—		y=uint8(zeros(48,1));
3	—		data0k = 0;
4	—		if Length $\sim$ = 48
5	_		return;
6			end
7	—		<pre>checksum=typecast(Data(1:4), 'int32');</pre>
8			if checksum ~= 1234567898
9			return;
10			end
11			
12	_		y = Data(1:48);
13			data0k = 1;

Using Python or C language, one can similarly listen for MAVLink's ACTUATOR\_CONTRO L\_TARGET message to obtain the "rfly\_ext" data. Readers are encouraged to try this out themselv es.

# 4.4. rfly\_insils.msg

Message Definition Format for Internal Data Transmission in PX4:

uint64 timestamp # time since system start (microseconds)
uint32 checksum # checksum/flag
int32[8] in\_sil\_ints # 8D int signals
float32[20] in\_sil\_floats # 20D float signals

The routine for this message is continuously being improved and can be broadly utilized in fo

ur aspects:

- Directly modifying the PX4 source code by adding subscription code for "rfly\_insils", a
  nd then publishing this message through Simulink to establish communication between t
  he Simulink controller and the modified code interface.
- Directly modifying the PX4 source code by adding publishing code for "rfly\_insils", an d then subscribing to this message through Simulink.
- Generating two automatic code applications for subscribing and publishing communicat ion between the two.
- Modifying the PX4 source code in two locations to enable bidirectional communication and connect the two sets of code.

# 5. Flight Log Recording and External Data Communication Interface

# 5.1. Simulation Ground Truth Data Analysis

#### 5.1.1. Offline Acquisition Method

For the i-th aircraft, you only need to create a new file under PX4PSP\CopterSim called Copt erSim+i+.csv (for example, CopterSim1.csv). After each simulation, it will record the simulation g round truth data (similar to RflySim3D received data, including position, velocity, motor RPM, et c.). For detailed operational steps, please refer to the following:\*\PX4PSP\RflySimAPIs\2.RflySim Usage\1.BasicExps\e14 Log-Get\Readme.pdf.

#### 5.1.2. Online Acquisition Method

RflySim provides two methods for simulation data retrieval and analysis:

- Method One: Detailed operational steps for the MATLAB/Simulink version are as follo ws:\*\PX4PSP\RflySimAPIs\10.RflySimSwarm\0.ApiExps\7.DataAnalysis\_Mat\Readm e.pdf。
- Method Two: Detailed operational steps for the Python version are as follows:\*\PX4PSP \RflySimAPIs\10.RflySimSwarm\0.ApiExps\8.DataAnalysis Py\Readme.pdf。

# 5.2. Flight Data Analysis

#### 5.2.1. Offline Log Analysis

Detailed operational steps for offline log analysis can be found at:\*\PX4PSP\RflySimAPIs\2.

 $RflySimUsage \label{eq:lbasicExps} e4\_Log-Reads-Python 38 Env \Readme.pdf_{\circ}$ 

#### 5.2.2. Online Log Analysis

Visit https://logs.px4.io/ and upload the ulog file for analysis.

#### 5.3. Controller and External Data Communication Interface

#### 5.3.1. Actuator output Message - HIL Simulation

Contents of the "actuator output.msg" file:

```
uint64 timestamp# time since system start (microseconds)uint8 NUM_ACTUATOR_OUTPUTS= 16uint8 NUM_ACTUATOR_OUTPUT_GROUPS= 4 # for sanity checkinguint32 noutputs# valid outputsfloat32[16] output# output data, in natural output units
```

This message is only used for hardware-in-the-loop (HIL) simulation.

#### 5.3.2. pwm\_output Message - HIL & Actual Flight

Contents of the "pwm output.msg" message file:

```
uint64 timestamp # Time since system start (microseconds)
uint64 error_count # Timer overcapture error flag (AUX5 or MAIN5)
uint32 pulse_width # Pulse width, timer counts
uint32 period # Period, timer counts
```

This message is only used for HIL and actual flight, and the flight controller's output must sup

port px4io.

#### 5.3.3. actuator\_control\_0—HIL & Actual Flight

Contents of the "actuator controls.msg" message file:

```
uint64 timestamp
                             # time since system start (microseconds)
uint8 NUM_ACTUATOR_CONTROLS = 8
uint8 NUM_ACTUATOR_CONTROL_GROUPS = 6
uint8 INDEX_ROLL = 0
uint8 INDEX_PITCH = 1
uint8 INDEX_YAW = 2
uint8 INDEX_THROTTLE = 3
uint8 INDEX_FLAPS = 4
uint8 INDEX_SPOILERS = 5
uint8 INDEX_AIRBRAKES = 6
uint8 INDEX_LANDING_GEAR = 7
uint8 INDEX GIMBAL SHUTTER = 3
uint8 INDEX_CAMERA_ZOOM = 4
uint8 GROUP_INDEX_ATTITUDE = 0
uint8 GROUP_INDEX_ATTITUDE_ALTERNATE = 1
uint8 GROUP_INDEX_GIMBAL = 2
```

```
uint8 GROUP_INDEX_MANUAL_PASSTHROUGH = 3
uint8 GROUP_INDEX_ALLOCATED_PART1 = 4
uint8 GROUP_INDEX_ALLOCATED_PART2 = 5
uint8 GROUP_INDEX_PAYLOAD = 6
uint64 timestamp_sample  # the timestamp the data this control response is based on was
sampled
float32[8] control
```

# TOPICS actuator\_controls actuator\_controls\_0 actuator\_controls\_1 actuator\_controls\_2 ac tuator\_controls\_3

# TOPICS actuator\_controls\_4 actuator\_controls\_5

# TOPICS actuator\_controls\_virtual\_fw actuator\_controls\_virtual\_mc

This message is primarily used to convey actuator control commands, with different message IDs corresponding to different control groups. Please refer to section 8.1 in the PX4 control group definition, such as the actuator\_control\_0 series messages. It corresponds to the control group of m ultirotors, supporting both hardware-in-the-loop simulation for multirotors and actual flight on real vehicles. Note: When using this message, it is necessary to select the option to suppress the actuat or\_control\_0 message in the installation script of RflySim (a simulator). Additionally, it cannot be directly mapped to motors; instead, the output needs to be configured using PX4's mixer rules.

#### 5.4. Flight Controller Communication Interface with External Data

Note: Please refer to the corresponding example code for this section:\*\PX4PSP\RflySimAPI s\5.RflySimFlyCtrl\0.ApiExps\9.PX4CtrlExternalTune。

#### 5.4.1. Port 20100 Series—Receiving Internal State Estimation Values from PX4

The 20100 series port primarily receives internal state estimation values from PX4. The received data includes:

```
struct outHILStateData{ // mavlink data forward from Pixhawk
        uint32_t time_boot_ms; //Timestamp of the message
        uint32_t copterID;
                             //Copter ID start from 1
        int32_t GpsPos[3];
                              //Estimated GPS position, lat&long: deg*1e7, alt: m*1e3 and up
is positive
        int32_t GpsVel[3];
                             //Estimated GPS velocity, NED, m/s*1e2->cm/s
        int32_t gpsHome[3]; //Home GPS position, lat&long: deg*1e7, alt: m*1e3 and up is p
ositive
        int32_t relative_alt; //alt: m*1e3 and up is positive
        int32_t hdg;
                              //Course angle, NED,deg*1000, 0~360
        int32_t satellites_visible; //GPS Raw data, sum of satellite
        int32_t fix_type; //GPS Raw data, Fixed type, 3 for fixed (good precision)
        int32_t resrveInit;
                                  //Int, reserve for the future use
        float AngEular[3]; //Estimated Euler angle, unit: rad/s
        float localPos[3]; //Estimated locoal position, NED, unit: m
        float localVel[3]; //Estimated locoal velocity, NED, unit: m/s
        float pos_horiz_accuracy; //GPS horizontal accuracy, unit: m
        float pos_vert_accuracy; //GPS vertical accuracy, unit: m
                               //float, reserve for the future use
        float resrveFloat;
```

}

The naming format for the receiving port of the i-th aircraft is: 20100 + (2\*i-1). For example,

for the first aircraft, the port is 20101, for the second aircraft, it is 20103, and so on.

# 5.4.2. Port 30100 Sesries—Receiving CopterSim Flight Simulation Values and Se nding rfly\_ctrl Messages to the Flight Controller

The 30100 series port receives flight simulation values from CopterSim and sends rfly\_ctrl m essages to the flight controller. The received data includes:

```
struct SOut2Simulator
{
   int copterID; //飞机 ID
   int vehicleType; //飞机构型
   double runnedTime; // 仿真时间
   float VelE[3]; //NED 地球系速度
   float PosE[3]; //NED 地球系位置
   float AngEuler[3]; //欧拉角
   float AngQuatern[4]; //四元数
   float MotorRPMS[8]; //电机转速 RPM
                     //机体轴加速度
   float AccB[3];
   float RateB[3];
                     //机体轴角速度
   double PosGPS[3];
                     //地球 GPS 经纬高
}
//SOut2Simulator 的定义与 Simulink 里面新加的 MavVehileInfo 结构体完全一致
//可以直接对新的 simulink 模型的 MavVehile3DInfo 输出口数据进行打包
//两个结构体的长度都是 152
sizeof(SOut2Simulator) = sizeof(MavVehileInfo) = 152
打包封装的结构体为
typedef struct _netDataShort
ł
   TargetType tg; //这里目前随便填, 写1即可 uint32
   int
            len; //这个长度为传输结构体长度,目前是152
             payload[192]; //这里面前 152 位存放了 SOut2Simulator 结构体数据, 后面的 40 位保留
   char
}netDataShort;
//UDP 包的总长度为 200
sizeof(netDataShort) = 200
```

//发送的 UDP 端口号为: 20010

The naming format for the receiving port of the i-th aircraft is: 30100 + (2\*i-1). For example,

for the first aircraft, the port is 30101, for the second aircraft, it is 30103, and so on.

The format of the uORB message data it sends is:

uint64 timestamp	<pre># time since system start (microseconds)</pre>
uint32 flags	# control flag
uint8 modes	# mode flag
float32[16] controls	# 16D control signals

The naming format for the sending port of the i-th aircraft is: 30100 + (2\*i-2). For example, f

or the first aircraft, the port is 30100, for the second aircraft, it is 30102, and so on.

#### 5.4.3. 40100 System Port—Receiving Internal rfly\_px4 Messages from Flight Co

#### ntroller

The 40100 system port receives internal rfly\_px4 messages from the flight controller. The rec eived data includes:

```
struct PX4ExtMsg {
    int checksum; //1234567898
    int CopterID;
    double runnedTime; //Current stamp (s)
    float controls[8];
}
```

The naming format for the receiving port of the i-th aircraft is: 40100 + (2\*i-1). For example, for the first aircraft, the port is 40101, for the second aircraft, it is 40103, and so on.

# 6. Code masking and replacement interface

When developing based on the underlying control algorithms of RflySim, to validate the developed control algorithms, we need to mask the output of the PX4 software. In most cases, we only need to directly mask the motor output in the PX4 software system. However, certain specific development tasks require masking a certain intermediate quantity of a module in the PX4 software system to meet development needs. For example, if we need to mask the module of the attitude angula r rate loop controller in the PX4 software system (this location is for PX4 version 1.12.3, for other versions, please refer to the PX4 official documentation) at: \*\PX4PSP\Firmware\src\modules\mc\_ rate\_control. Open the "MulticopterRateControl.cpp" file in this folder. Based on the PX4 source c ode architecture, we can know that the output uORB message of the attitude angular rate loop is "a ctuator\_controls\_0" (detailed definition of this message can be found at https://docs.px4.io/v1.12/e n/concept/mixing.html). By examining the code, we can find that the code for publishing the "actu ator\_controls\_0" message is as follows (you can also find it by searching for "\_actuators\_0\_pub.pu blish(actuators);"):



To mask the above two lines of code, there are two methods that can be used.

**Method 1:** We only need to delete them, comment them out, or replace them with other invali d code. Since PX4's compilation checks are very strict, directly commenting out the above two line

s of code may lead to compilation errors due to the actuators being defined but not used. Therefore, here we need to use the UNUSED macro to achieve masking. The masking of the code can follow one of the following rules depending on the situation.

- a) Replace "\_actuators\_0\_pub.publish(actuators);" with "" (empty string, equivalent to d eletion. Note that this method is only applicable when there will be no error due to unuse d variable.)
- b) Replace "\_actuators\_0\_pub.publish(actuators);" with "//\_actuators\_0\_pub.publish(act uators);" (this is equivalent to commenting out the line. Note the precautions mentioned above.)
- c) Replace "\_actuators\_0\_pub.publish(actuators);" with "UNUSED(actuators);" (this is equivalent to replacing it with an invalid statement. Note that this method is applicable w hen directly deleting the actuators variable would result in an error, and it is suitable for PX4 1.12 and earlier versions, as the UNUSED macro was removed starting from firmw are version 1.13.)
- d) Replace "\_actuators\_0\_pub.publish(actuators);" with "(void) (actuators);" (this is equ ivalent to replacing it with an invalid statement. Note that this method is applicable whe n directly deleting the actuators variable would result in an error, and it is suitable for all versions including firmware version 1.13.)

**Method 2:** Another approach is to directly replace the file to be modified with a pre-modifi ed file. The platform's provided interfaces can meet the requirements mentioned above. The one-cl ick installation script of the RflySim platform offers a file replacement feature. The core idea is to describe the files and their contents to be replaced according to a given Excel file template. When r unning the installation script, you input the address of the Excel file, enabling automatic file replac ement or modification of file contents during platform installation. For specific masking methods, please refer to the example files for details: <u>2.AdvExps\e0\_AdvApiExps\1.CusMaskPX4Code\Rea</u> <u>dme.pdf</u>

# 7. Multi-module parallel development interface

The latest version of the RflySim platform supports the rapid creation of multiple modules for parallel development. Based on the multi-process running state in the PX4 software system, the P X4 application generated automatically by MATLAB's code generation is named "px4\_simulink\_a pp." You can rename the PX4 application using the "MATLAB Command Line Interface." By rena ming px4\_simulink\_app, you can continue to build models through Simulink to generate another a pplication with the name px4\_simulink\_app. If you need to add another application, you can continue to rename it, and so on. In theory, this approach can facilitate the creation of numerous PX4 applications to meet development needs. For detailed usage of the interface, please refer to the experi

ment file: \*\RflySimAPIs\5.RflySimFlyCtrl\2.AdvExps\e0\_Basic-Interface\4.MultPX4App\Read me.pdf.

# 8. Different aircraft model development interface

# 8.1. Introduction to PX4 Flight Controller

The PX4 architecture ensures that no special handling for airframe layout is required within th e core controller. Mixer refers to the distribution of input commands (e.g., right turn from the remo te controller) to the actuators of motors and servos (such as electronic speed controllers or servo P WM commands). For fixed-wing aircraft, for instance, where each aileron is controlled by a servo, mixing involves controlling one aileron to lift while the other drops. Similarly, for multirotors, pit ch operation requires changing the speed of all motors. Separating the mixing logic from the actual attitude controller significantly improves reusability.

A specific controller sends a normalized force or torque command (scaled to -1..+1) to the mi xer, which then sets each individual actuator accordingly. Control output drivers (such as UART, U AVCAN, or PWM) then translate the mixer's output into the native units for the actuators' operatio n, such as outputting a PWM command with a value of 1300.



The control channel outputs of PX4 mainly consist of 4 control groups, which are:

actuator\_controls\_0:The primary control channels of the flight controller are used to output c ontrol signals for various channels such as pitch, roll, yaw, throttle, etc. Their specific definiti ons are as follows:

```
Control Group #0 (Flight Control)
    0: roll (-1..1)
    1: pitch (-1..1)
    2: yaw (-1..1)
    3: throttle (0..1 normal range, -1..1 for variable pitch / thrust reversers)
    4: flaps (-1..1)
    5: spoilers (-1..1)
    6: airbrakes (-1..1)
    7: landing gear (-1..1)
```

actuator\_controls\_1:Backup control channels, used in VTOL for outputting control signals in

fixed-wing mode. Their specific definitions are as follows:

Control Group #1 (Flight Control VTOL/Alternate)

• 0: roll ALT (-1..1)

```
• 1: pitch ALT (-1..1)
```

```
• 2: yaw ALT (-1..1)
```

- 3: throttle ALT (0..1 normal range, -1..1 for variable pitch / thrust reversers)
- 4: reserved / aux0
- 5: reserved / aux1
- 6: reserved / aux2
- 7: reserved / aux3
- > actuator\_controls\_2: Gimbal control channel. Their specific definitions are as follows:

Control Group #2 (Gimbal)

•	0:	gimbal roll
•	1:	gimbal pitch
•	2:	gimbal yaw
•	3:	gimbal shutter
•	4:	reserved
•	5:	reserved
•	6:	reserved

- 7: reserved (parachute, -1..1)
- > actuator\_controls\_3: Remote control mapping channel. Their specific definitions are as follo

ws:

Cont	rol	Group #3 (Manual Passthrough)
•	0:	RC roll
•	1:	RC pitch
•	2:	RC yaw
•	3:	RC throttle
•	4:	RC mode switch
•	5:	RC aux1
•	6:	RC aux2
•	7:	RC aux3

#### 8.2. PX4 Mixer Definition

One of the functions of the mixer in PX4 is to connect the outputs of upper-level application modules (such as the output from the attitude controller algorithm) with the actuators of the underlying hardware (corresponding to PWM values for motors or servos). The mixer module effectively isolates the upper-level application modules from the hardware interface, so developers don't need to worry about which motor receives which control signal when writing upper-level application modules. Additionally, by changing different mixer configuration files, PX4 can adapt to different ai rframe types without needing to modify the code of upper-level application modules. The specific code can be found in \*\PX4PSP\Firmware\ROMFS\px4fmu\_common\mixers.

The default settings for mixer files in the PX4 software system can be found in the \*\PX4PSP \Firmware\ROMFS\px4fmu\_common\init.d\airframes folder. Alternatively, these settings can be o verridden by placing a mixer file with the same name in the \*/etc/mixers/ directory on the SD card. Mixer files with the prefix "xxxx.main.mix" are mapped to the main (MAIN) output, while files n amed "yyyy.aux.mix" are mapped to the auxiliary (AUX) output. The prefixes (xxxx/yyyy) depend on the airframe and its configuration. Typically, MAIN and AUX outputs correspond to MAIN an d AUX PWM outputs, but when enabled, these may be loaded onto the UAVCAN (or other) bus. The main mixer file name (prefix xxxx) is used in the airframe configuration with the comma nd "set mixer xxxx," for example, calling "set mixer quad\_x" in the airframes/4011\_dji\_f450 confi guration loads the main mixer file quad\_x.MAIN.mix). The AUX mixer file (prefix yyyy) depends on the airframe setting or defaults, such as:

- MIXER\_AUX can be used to set which yyyy.aux.mix file is loaded. For example, in the file \* \PX4PSP\Firmware\ROMFS\px4fmu\_common\init.d\airframes\13006\_vtol\_standard\_d elta\*, after set MIXER\_AUX vtol\_delta, the loaded mixer file will be \*\PX4PSP\Firmware\ ROMFS\px4fmu\_common\mixers\vtol\_delta.aux.mix.
- For multirotor and fixed-wing aircraft models, if the MIXER\_AUX option is not set, the defa ult loaded mixer file will be \*\PX4PSP\Firmware\ROMFS\px4fmu\_common\mixers\pas s.aux.mix. (Note: `pass.aux.mix` is the pass-through mixer file for the remote control, w hich passes the values of four user-defined remote control channels (set using `RC\_MAP AUXx`/`RC\_MAP\_FLAPS` parameters) to the first four outputs of the AUX output.)
- Vertical take-off and landing (VTOL) drones will load the `yyyy.aux.mix` set if relevant settings are applied; otherwise, only the mixer file set by MIXER will be loaded.
- Enabling the general control mode (with the output mode set to AUX) will override the M
   IXER\_AUX setting in the airframe and load the AUX output on \*\PX4PSP\Firmware\ROM
   FS\px4fmu\_common\mixers\mount.aux.mix\*.

Note: The loading of mixer files as described above is implemented through the file \*\PX4PS P\Firmware\ROMFS\px4fmu common\init.d/rc.interface.

# 8.3. The syntax of PX4 mixer files

Mixer files are text files that define one or more mixer definitions: a mapping between one or more inputs and one or more outputs. There are mainly four types of definitions: multirotor mixer, helicopter mixer, summing mixer, and null mixer.

- $\blacktriangleright$  multirotor mixer: defines the outputs of a + or x-shaped rotorcraft with 4, 6, or 8 outputs.
- helicopter mixer: defines the outputs of helicopter swashplate servos and main motor ESCs (t he tail rotor is a separate mixer).
- summing mixer: combines zero or more control inputs into a single actuator output. The inputs are scaled, and the mixing function sums the results before applying the output scaler.
- > null mixer: produces an output of zero for the actuator output (when not in fail-safe mode).

The output produced by each mixer depends on the type and configuration of the mixer. For e xample, a multirotor mixer can have 4, 6, or 8 outputs depending on the aircraft type, while a sum ming mixer or null mixer only produces one output. Multiple mixers can be specified in each file. The output order (assigning mixers to actuators) is specific to the device that reads the mixer defin itions, and for PWM, the output order matches the declaration order.

The first statement defined in each mixer file is:

<tag>: <mixer arguments>

The tag here represents the selected mixer type, as follows:

R: Multirotor mixer

- H: Helicopter mixer
- M: Summing mixer
- Z: Null mixer

#### 8.4. Summing Mixer—Additive Mixer

The Summing Mixer is used to control UAV actuators and servos. It combines zero or more c ontrol inputs into a single actuator output. The inputs are scaled, and the mixing function sums the results before applying the output scaler. The minimum actuator traversal time limit can also be sp ecified in the output scalar (inverse of rotation rate). A simple mixer definition is as follows:

M: <control count>

0: <-ve scale> <+ve scale> <offset> <lower limit> <upper limit> <traversal time>

If <control count> is zero, then the sum is effectively zero, and the mixer outputs a fixed value constrained by <lower limit> and <upper limit>.

The second line defines the output scalar using the scalar parameters mentioned above. Altho ugh calculations are performed as floating-point operations, values stored in the definition file are scaled by a factor of 10,000; for instance, an offset of -0.5 is encoded as -5000. The <traversal tim e> on the output scale (optional) is used for actuators, and if the value is too large, it may damage t he aircraft—e.g., a tilted actuator on tilt-rotor VTOL aircraft. It can be used to limit the rate of cha nge of actuators (if not specified, rate limiting should not be applied). For example, a <traversal t ime> value of 20,000 will limit the rate of change of the actuator, requiring at least 2 seconds to go from <lower limit> to <up>
typer limit> and vice versa.

Note 1: <traversal time> should only be used when required by hardware!

Note 2: Do not impose any restrictions on actuators controlling vehicle attitude (such as servo s for aerodynamic surfaces), as this can easily lead to controller instability.

Proceed with defining the inputs and their scaling for <control count>, in the following form:

S: <group> <index> <-ve scale> <+ve scale> <offset> <lower limit> <upper limit>

Note 3: s: must be below o:.

Note 4: Mixer outputs (where <group>=0 and <index>=3) with throttle input will not operate in t he armed or pre-armed state. For example, a servo with four inputs (roll, pitch, yaw, and throttle) w ill not move even with roll/pitch/yaw signals in the armed state.

The <group>value identifies the control group from which the scalar mixer reads, while the <in dex> value indicates the offset within that group. These values are specific to the devices defined in the mixer. When used for mixing vehicle controls, mixer group 0 is for vehicle attitude control, wi th 0 to 3 typically representing roll, pitch, yaw, and thrust respectively. The remaining fields utilize the parameter configurations for controlling scalers as discussed above. Values stored in the defini

tion file are scaled by a factor of 10,000 when computed as floating-point operations; for example, an offset of -0.5 is encoded as -5000. Below is an explanation of a typical mixer file example. For detailed example analysis, please refer to:<u>https://docs.px4.io/v1.13/en/dev\_airframes/adding\_a\_ne</u> w frame.html#mixer-file。

#### 8.5. Null Mixer—Flight Controller

This mixer does not consume any control channels and generates a single actuator output with a constant value of zero. Typically, the Null Mixer is used as a placeholder in mixer collections to achieve a specific pattern of actuator outputs. It can also be utilized to control the output values for fault safety mechanisms (output is 0 during normal operation; during fault safety, the mixer is ign ored, and the fault safety values are used instead). The definition is as follows:

#### Ζ:

#### 8.6. Multirotor Mixer—Multirotor Mixer

The Multirotor Mixer combines four control inputs (roll, pitch, yaw, thrust) into a set of actua tor outputs used for driving motor speed controllers. The definition is as follows:

R: <geometry> <roll scale> <pitch scale> <yaw scale> <idlespeed>

Supported aircraft types include:

- 4x Quadcopter X Configuration
- ➢ 4+ Quadcopter Plus Configuration
- ➢ 6x Hexacopter X Configuration
- 6+ Hexacopter Plus Configuration
- ➢ 8x Octocopter X Configuration
- ➢ 8+ Octocopter Plus Configuration

The proportional values for roll, pitch, and yaw determine the proportion of roll, pitch, and yaw control relative to thrust control. When computed as floating-point operations, values stored in the definition file are scaled by a factor of 10,000; for example, 0.5 is encoded as 5000. The range of finputs for roll, pitch, and yaw is from -1.0 to 1.0, while the range for thrust input is from 0.0 to 1.0. The output range for each actuator is from -1.0 to 1.0.

The idle speed ranges from 0.0 to 1.0. Idle speed is relative to the maximum speed of the mot or, which is the speed at which the motor is commanded to rotate when all control inputs are zero. In the case of actuator saturation, the values for all actuators are readjusted to constrain the saturate d actuators to 1.0.

#### 8.7. Helicopter Mixer—Helicopter Mixer

The Helicopter Mixer combines three control inputs (roll, pitch, thrust) into four outputs (coll ective and main motor ESC settings). The first output of the helicopter mixer is the throttle setting

for the main motor. The subsequent outputs are for the servo of the swashplate. Tail rotor control c an be achieved by adding a simple mixer. The thrust control input is used for the main motor settin g and collective pitch of the swashplate. It employs a throttle curve and a pitch curve, each compos ed of five points

Note: The throttle and pitch curves map the positions of the "throttle" stick input to throttle va lues and pitch values (respectively). This allows for flight characteristics to be adjusted for differen t types of flight.

The Helicopter Mixer is defined as follows:

H: <number of swash-plate servos, either 3 or 4>

T: <throttle setting at thrust: 0%> <25%> <50%> <75%> <100%>

P: <collective pitch at thrust: 0%> <25%> <50%> <75%> <100%>

T: Defines points for the throttle curve. P: Defines points for the pitch curve. Both curves cons ist of 5 points ranging from 0 to 10000. For a simple linear variation, the five values of the curve s hould be 0, 2500, 5000, 7500, 10000.

The definition for each servo of the swashplate (3 or 4) is as follows:

S: <angle> <arm length> <scale> <offset> <lower limit> <upper limit>

<angle> is in degrees, with 0 degrees representing the direction of the nose. Positive angles are clockwise. <arm length> is a normalized length, where 10000 equals 1. If all servo arms are of the s ame length, the value should be 10000 for each. A larger arm length will decrease the amount of se rvo deflection, while a shorter arm length will increase it. The servo output is scaled by <scale> / 10000. After scaling, <offset> is applied, and its value should be between -10000 and +10000. With in the full servo range, <lower limit> and <upper limit> should be -10000 and +10000, respectivel y.

The tail rotor can be controlled by adding a Summing Mixer:

M: 1

S: 0 2 10000 10000 0 -10000 10000

By directly mapping the tail rotor to the yaw command. This applies to both tail rotors control led by two servos and tail rotors with a dedicated motor.

The 130-blade helicopter mixer file is as follows:

```
H: 3
т:
       0
           3000
                   6000
                         8000 10000
     500
           1500
                   2500
                         3500
                                4500
P:
# Swash plate servos:
S:
       0 10000 10000
                            0 -8000
                                       8000
S:
     140 13054 10000
                            0 -8000
                                       8000
S:
     220 13054 10000
                            0 -8000
                                       8000
# Tail servo:
M: 1
                         0 -10000 10000
S: 0 2 10000 10000
```

• At 50% thrust, the slope of the throttle curve is slightly steep, reaching 6000 (0.6).

• At 100% thrust, it reaches 10000 (1.0) with a smaller slope.

- The pitch curve is linear, but its entire range isn't used.
- At 0% throttle, the collective control stick setting is already at 500 (0.05).
- At maximum throttle, the collective control stick is only at 4500 (0.45).
- Using higher values for this type of helicopter will cause the blades to stall.
- The swashplate system of this helicopter is set at angles of 0, 140, and 220 degrees.
- Servo arm lengths are not equal.
- Compared to the first servo system, the arm lengths of the second and third servo system s are 1.3054.
- The servos are limited to -8000 and 8000 because of mechanical constraints.

#### 8.8. VTOL Mixer—Vertical Takeoff and Landing (VTOL) Drone Mix

#### er

The vertical take-off and landing system utilize a multirotor mixer as the output in multirotor mode and a sum mixer as the output in fixed-wing mode. The mixer system of the vertical take-off and landing UAV can be either combined into a single mixer, where all actuators are connected to IO or FMU ports, or divided into separate mixer files for IO and AUX.

# 9. PX4 Native Interfaces

<u>PX4</u> is highly portable and serves as an operating system-independent solution for unmanned aerial vehicles (UAVs). Developed by world-class developers from both industry and academia, it benefits from active global community support, powering a wide range of vehicles from racing and cargo drones to ground vehicles and underwater vehicles. For more information, visit the official website: <u>https://docs.px4.io/main/en/</u>. Below are some commonly used uORB messages, modules, and parameters in the PX4 software system.

#### 9.1. Getting Started with the PX4 Software System

Drones are unmanned "robotic" vehicles that can be operated remotely or autonomously, and they are widely used in many consumer, industrial, governmental, and military applications. These applications include but are not limited to aerial photography/videography, cargo transportation, ra cing, search and surveying, among others. Different types of drones are used in aerial, ground, mar ine, and underwater environments. These drones are referred to as Unmanned Aerial Vehicles (UA Vs), Unmanned Aerial Systems (UAS), Unmanned Ground Vehicles (UGV), Unmanned Surface V ehicles (USV), and Unmanned Underwater Vehicles (UUV).

The "brain" of a drone is known as the autopilot system. It runs flight stack software on the ve hicle controller ("flight controller") hardware. Some drones are also equipped with a separate onbo ard computer, providing a powerful platform for networking, computer vision, and many other task PX4 is a powerful open-source drone flight control stack, with some key features including:

- Control a variety of vehicle frames/types, including drones (multirotors, fixed-wing aircraft, a nd VTOL aircraft), ground vehicles, and underwater vehicles.
- Offer a robust selection of vehicle controllers, sensors, and other peripheral devices.
- Provide flexibility and robustness in flight modes and safety features.
- Deeply integrate with onboard computers and robot APIs (such as ROS 2, MAVSDK). PX4 serves as a core component of a broader drone platform, which includes QGroundContro l-based ground stations, Pixhawk hardware, and MAVSDK, integrating with companion com puters, cameras, and other hardware using the MAVLink protocol.

The ground control station for Dronecode is called QGroundControl. You can use QGroundC ontrol to flash PX4 onto vehicle control hardware, configure vehicles, change various parame ters, get real-time flight information, and create and execute fully autonomous missions. QGr oundControl runs on Windows, Android, MacOS, or Linux. Download and install it from her



PX4 supports aerial, ground, and underwater vehicles. You can view all types and variants ("f rames") of vehicles for PX4 testing/tuning here: Airframe Reference. Choose different vehicles bas ed on the type you need:

**Multirotors:** Offer precise hovering and vertical takeoff but have shorter range and typically fly slower. They have modes in PX4 that make them easy to fly and are the most popular type of fl ying vehicle.

Helicopters: Similar to multirotors but mechanically more complex and efficient.

s.

**Fixed-wing aircraft:** Provide longer flight times and faster speeds, making them better for ap plications like ground surveys. However, they are more challenging to fly and land than multirotor s. They are not suitable if you need to hover or fly at very slow speeds (e.g., when surveying vertic al structures).

**Vertical Takeoff and Landing (VTOL) drones:** Come in various types such as tiltrotors, tail sitters, quadplanes, etc. They offer the dual advantage of vertical takeoff like multirotors and then t ransitioning to forward flight like airplanes. They are typically more expensive and harder to manu facture and tune than multirotors and fixed-wing aircraft.

**Balloons/Airships:** Are lighter-than-air flying vehicles that typically offer high-altitude, long -duration flights, usually with limitations (or no limitations) on range and speed control.

Rovers are ground vehicles similar to cars. They are easy to control and often fun.

Unmanned boats: Are surface vehicles.

Unmanned underwater vehicles: Are underwater vehicles.



#### 9.1.1. Firmware Download

After installing the RflySim platform, you can find the complete PX4 source code at the path: \*\PX4PSP\Firmware. Open the WSL subsystem on the desktop (\*\Desktop\RflyTools\Win10WSL. lnk) for compilation. After the compilation is complete, open the QGC software for firmware down load. Note: You can also directly download the firmware through QGC if there is an internet conne ction.

#### 9.1.2. Model Configuration

After firmware flashing is complete, launch the QGroundControl software. Once connected to the flight controller, select the "Q" icon > Vehicle Setup > Airframe (sidebar) to open the airframe configuration. Choose the vehicle group/type that matches your frame, and then within the group, use the drop-down menu to select the frame most suitable for your vehicle. For example, if you are simulating quadcopter hardware in a loop, you can select the frame as follows: Simulation -> HIL Quadcopter X.



#### 9.1.3. Hardware-in-the-loop (HIL) simulation

Connect the flight controller to the computer, open the QGroundControl ground station, and s elect the airframe as follows: Simulation -> HIL Quadcopter X. In the safety options, set: Safety -> Hardware-in-the-loop Simulation -> HITL enabled.



Open the "\*\Desktop\RflyTools\HITLRun.lnk" to start the hardware-in-the-loop simulation sc ript with a single click. Input the port number of the flight controller and wait until the message ba r at the bottom left of CopterSim displays: "PX4: GPS 3D fixed & EKF initialization finished". Th en, you can unlock and take off the aircraft in QGC.

QGroundControl Daily					- 0	×
Back < 😵 Vehicle Setup						
▲ 概況			30.0 m			
■▲ 同件		<ul> <li>● 立即著陆</li> </ul>				
***	1 A 4	● 留待但不着陆				
机架	_ ♠ ♣ ‴	<ul> <li>         ·        ·</li></ul>				
200 或放展		留待时间	0.0 s			
AULLAF		留待高度	10.0 m			
100 飞行模式						
	着陆模式设置					
***2**						
<b></b> чл	<u>چ</u>	着陆下降速率: ✓ 几秒后错定。	0.7 m/s			
Ph #4		, 10 / H M/CT				
	无线数传日志					
PID Tuning						
A Light Pohysion		铺什就我口志到飞机里				
			external HITL			
▲ 相机	硬件在环仿吉		HITL and SIH disabled			
			HITL enabled			
参数 参数			SIH enabled			
		已启用HITL:	HITL enabled	×		



#### 9.1.4. Aircraft actual flight

Choose the airframe for your specific vehicle, for example: for a quadcopter, you can select th e airframe as follows: Quadrotor X -> DJI F450 w/ DJI ESCs. In the safety options, set: Safety -> Hardware-in-the-loop Simulation -> external HITL. For more detailed steps on actual flight, please refer to the experiment: <u>1.BasicExps\e5-AttitudeCtrl\e5.4\Readme.pdf</u>.

# 9.2. PX4 Official Flight Controller Support Introduction

The flight controllers officially supported by PX4 are maintained by the PX4 Autopilot maint ainers and the Dronecode team. They comply with the Pixhawk standard. For more up-to-date info rmation, please visit: <u>https://docs.px4.io/main/en/flight\_controller/</u>. The following table lists some commonly used flight controller hardware and compilation commands:

Serial	Onboard Informatio		
Numb	n	Hardware Name	Compilation Command
er			
1		CUAV Pixahwk V6X (FMUv6X)	
2	FMUv6X and FMU	Holybro Pixhawk 6X (FMUv6X)	px4_fmu-v6x_default
3	v6C	Holybro Pixhawk 6C (FMUv6C)	
4		Holybro Pix32 v6 (FMUv6C)	px4_fmu-v6c_default
5	EMIL-5 and EMIL-	Pixhawk 4 (FMUv5)	
6	5X (STM32F7, 201	Pixhawk 4 mini (FMUv5)	
7		CUAV V5+ (FMUv5)	px4_fmu-v5_default
8	9/20)	CUAV V5 nano (FMUv5)	
9	FMUv4 (STM32F	Pixracer	px4_fmu-v4_default
10	4, 2015)	Pixhawk 3 Pro	px4_fmu-v4pro_default
11	EMILy2 (STM22E	Pixhawk 2	
12	1 2014)	Pixhawk Mini	px4_fmu-v3_default
13	4, 2014)	CUAV Pixhack v3	
14	FMUv2 (STM32F 4, 2013)	Pixhawk	px4_fmu-v2_default

# 9.3. Introduction to Commonly Used uORB Messages in PX4

Seria 1 Nu mber	Name	Description	File Address
1	actuator_controls.msg	Driver control signal	*\PX4PSP\Firmware\msg\actuato r_controls.msg
2	actuator_outputs.msg	Driver output signal	*\PX4PSP\Firmware\msg\actuato r_outputs.msg
3	3 input_rc.msg	The remote control inpu	*\PX4PSP\Firmware\msg\input_r
		ts a message	c.msg
4	led_control.msg	Controls single or multi	*\PX4PSP\Firmware\msg\led_co
		ple external LEDs	ntrol.msg
5	vehicle_status.msg	Vehicle status message	*\PX4PSP\Firmware\msg\vehicle
			_status.msg
6	vehicle_imu.msg	Message output from ve	*\PX4PSP\Firmware\msg\vehicle
		hicle IMU	_imu.msg

For more uORB message descriptions, please refer to:<u>https://docs.px4.io/main/en/msg\_docs/</u>

# 9.4. Introduction to Commonly Used Modules in PX4

Seria 1 Nu mber	Name	Description	File Address
1	mc_rate_control	Multi-rotor speed control module	*\PX4PSP\Firmware\src\modules \mc_rate_control
2	mc_pos_control	Multi-rotor position control module	*\PX4PSP\Firmware\src\modules \mc_pos_control

3	mc_att_control	Multi-rotor attitude control module	*\PX4PSP\Firmware\src\modules \mc_att_control
4	navigator	Navigation control module	*\PX4PSP\Firmware\src\modules \ navigator

For more uORB message descriptions, please refer to: https://docs.px4.io/main/en/

# 9.5. Introduction to Commonly Used Parameters in PX4

序号	Name	Description
1	MPC_THR_MIN	Minimum thrust in automatic thrust control
2	MPC_THR_HOVER	Hover thrust
3	MPC_USE_HTE	Hover Thrust Source Selector
4	MC_ROLLRATE_P	Proportional gain of roll rate
5	MC_ROLLRATE_I	Integral gain of roll rate
6	MC_ROLLRATE_D	Differential gain of roll rate
7	MC_ROLLRATE_FF	Feed forward of roll acceleration
8	MC_ROLLRATE_K	Roll Rate Controller Gain, Global Gain of Controller

For more uORB message descriptions, please refer to:<u>https://docs.px4.io/main/en/</u>

# **10. Reference Materials**

[1]. 全权,杜光勋,赵峙尧,戴训华,任锦瑞,邓恒译.多旋翼飞行器设计与控制[M],电子工业出版 社,2018.

Quan Quan, Du Guangxun, Zhao Zhiyao, Dai Xunhua, Ren Jinrui, translated by Deng Heng, "Design and Control of Multirotor Aircraft" [M], Published by Electronic Industry Press, 201 8.

[2]. 全权,戴训华,王帅.多旋翼飞行器设计与控制实践[M],电子工业出版社,2020.

Quan Quan, Dai Xunhua, Wang Shuai, "Design and Control of Multirotor Aircraft: Practice" [M], Published by Electronic Industry Press, 2020.

# **11.Common Questions and Answers**

# 11.1.MATLAB/Simulink During automatic code generation, the follo

wing error is sometimes reported.



#### Caused by:

```
Validation error(s):
### Validating other build tools ...
Unable to locate build tool "Pixhawk Toolchain C Compiler": echo
Unable to locate build tool "Pixhawk ToolchainC Pre-Linker": echo
Unable to locate build tool "Pixhawk Toolchain Archiver": echo
```

In case of compilation errors, possible compilation problems can be classified as: MATLAB model problems, PX4 firmware problems, MATLAB model and PX4 firmware linking problems.

#### **Questions and answers:**

Dealing with MATLAB model problems. MATLAB automatic code generation checks the mo del at the initial stage of compilation, so these kinds of problems are often displayed within second s of clicking the compile button. The most common MATLAB problem is that the data of each inte rface does not match. Click the module that prompts the error to jump to the problem.



Address the PX4 firmware issue. If the PX4 source code has compilation problems, it will ge nerally be displayed in the compilation log prompt of MATLAB. The following figure shows the l ocation of the problem code. Modify it according to the prompt. The firmware for the PX4 in the p latform is located at PX4PSP \ Firmware.



Handles linking of MATLAB models to PX4 firmware. This kind of problem is often caused by the change of some interfaces due to the upgrade of PX4 firmware version, and the interfaces g enerated by MATLAB automatic code may not match, so errors will occur in the final link stage. T his kind of problem cannot see the specific error in MATLAB. You need to open Win10WSL (refer to other tools if other compilation tools are selected) and re-execute the following compilation co mmands, such as make px4\_fmu-v6c\_default (change other versions to their corresponding co mmands) to see the specific problem.



注意: 在定位飞控编译问题时, 应该保证半台是止确安装的, 代码版本和编译命令能够 相匹配。

# 11.2.In the automatic code generation controller, the delay module is used to directly generate control instructions, which causes the ai

# rcraft to fly around.

## Description of the problem:

Routine: <u>1. BasicExps\e5-AttitudeCtrl\Readme.pdf</u> disconnect the remote controller inputs C H1-CH4 from the inputConditioning module, and change the input signals of roll (CH1), pitch (CH 2), throttle (CH3) and yaw (CH4) to the desired fixed value inputs. Only the AttitudeControl \_ HI L. Slx is modified in this place, and then the hardware-in-the-loop simulation is carried out with th e same operation steps as the original AttitudeControl \_ HIL. Slx routine. After the hardware-in-th e-loop is started, the aircraft itself flies on the ground, and an error will be reported after a while. W hat is the reason? How to solve it?

修改前:



修改后:



Simulation phenomenon: After the change, the UAV flies on the ground during the hardware-in-the -loop simulation.



This modification is not feasible because the total delay of 200 steps is only 200 \* 0.001 = 0.2 seco nds. Because the px4 \_ simulink \_ app program starts to run when it is powered on, it is equivalent to inputting control instructions after powering on, and it becomes a full throttle state. In this case, the flight control has not been initialized, the filter triggers the divergent state, and the accurate po se information is not obtained.



Perfect solution: directly read the uORB message of the EKF state, judge that the filter is initialize d, and then delay a period of time to give control instructions. Simple solution: extend the delay ti me a little more, or directly use MATLAB function to write a trigger mechanism, and then give co ntrol instructions after the simulation time reaches 60s.

# 11.3.SIL Or HIL When simulating, RflySim3D Appear Fatal error:

# [File:D://Build/++UE4....]... Report an error

The specific error reporting interface is as follows:



#### **Questions and answers:**

The above RflySim3D error may be due to the compatibility problem caused by the graphics card driver of the computer. It is recommended to upgrade the graphics card driver to the latest ver sion and see if it can be solved. If it cannot be solved, please contact the relevant after-sales person nel of the RflySim platform.

#### 11.4. How to do UAV attitude autonomous control?

#### **Description of the problem:**

The detailed description is: we need to let the UAV fly to a certain height and fix it, and then l et the UAV move autonomously according to the input of the expected attitude angle given by ours elves, so that the three attitude angles can move to the given expected value.

#### **Questions and answers:**

The channel of the remote controller can be used to transmit trigger information. For example, when CH5 is dialed to the full position, the program starts to execute automatically. Write a state machine in Simulink to let the aircraft take off first, and then control the attitude after reaching the altitude. The specific design method of the controller belongs to the category of Simulink program ming, which can be understood by referring to relevant literature. If you want to input the informat ion such as the expected attitude angle from the outside after switching the fixed height, please use <u>0. ApiExps\9.PX4CtrlExternalTune\Readme.pdf</u> the external control interface in this directory to s ubscribe to the rfly \_ ctrls message (transmitted from the external program) in the flight control as the expected attitude angle.

#### 11.5. How to get the result data of attitude control hardware in the loo

### p? Do I know how to download the flight log to get what I want?

#### **Questions and answers:**

Please use <u>0. ApiExps\9.PX4CtrlExternalTune\Readme.pdf</u> the interface of the routine to obta in the real attitude (simulation value) of the model in Simulink in real time and the data from the fl ight control (fill in the flight control attitude). Similar data can also be obtained from the flight log.

# 11.6.QGC Yes Analyze Tools- Flight log, after refreshing when down loading, I can't find the log of the time corresponding to the hard ware in the ring.

#### **Questions and answers:**

Open the QGC and enter the vehicle setup



Go to the parameter label at the bottom, search for "log", find the parameter of the SDLOG \_ MODE, change it to the following figure, record the log from power-on to power-off, and then you can see the log.

🗢 Vehicle Setup	r	参数编辑器 取消 保存
搜索: log	清除 只易	from boot until shutdown 👻 Reset To Default
CBRK_FLIGHTTERM	121212	disabled stop logging. By
COM_FLT_PROFILE	Default	when armed until disarm (den arming the system,
GPS_DUMP_COMM	Disable	from boot until disarm
SDLOG_BOOT_BAT		depending on ALIX1 PC chart
SDLOG_DIRS_MAX		Vehicle reboot required after change
SDLOG_MISSION	Disabled	警告:在飞机飞行时修改值可能导致飞机不稳定,也可
SDLOG_MODE	from boot until disarm	能造成飞机飞丢。确保你知道你在做什么,并在保存之 前仔细检查你设置的值!
SDLOG_PROFILE	131	高级设置
SDLOG_UTC_OFFSET		
SDLOG_UUID		
SENC CDC DDIME		

# 11.7. Win10WSL When compiling the firmware, displays: region `A

# XI\_SRAM' overflowed by 15401072 bytes

**Description of the problem:** 



#### **Questions and answers:**

The problem is that the compiled firmware is larger than the content of the flight control, caus ing the firmware to overflow and report an error. You can enter the Cmake file of PX4 to comment out some unused modules. The specific address of Pixhawk series flight control is: \* PX4PSP \ Fir mware \ boards \ px4. For example, the experiment conducted is a quadrotor related low-level cont rol algorithm development experiment, the flight control used is Pixhawk 6C, and the unused mod ules in C:  $PX4PSP \setminus Firmware \setminus boards \setminus px4 \setminus fmu-v6c \setminus default$ . Cmake can be annotated (as foll ows). And then compile.

```
MODULES
airspeed_selector
px4_simulink_app
attitude_estimator_q
camera_feedback
commander
dataman
```

#### ekf2

esc\_battery events flight\_mode\_manager 注释掉固定翼姿态控制模块 # fw\_att\_control fw\_pos\_control\_l1 gyro\_calibration gyro\_fft land\_detector landing\_target\_estimator load\_mon local\_position\_estimator logger mavlink mc\_att\_control mc\_hover\_thrust\_estimator mc\_pos\_control mc\_rate\_control #micrortps\_bridge navigator rc\_update rover\_pos\_control sensors sih temperature\_compensation 
 # uuv\_att\_control
 注释掉 UUV 姿态控制模块

 # uuv\_pos\_control
 注释掉 UUV 位置控制模块
 vmount **# vtol\_att\_control** 注释掉 VTOL 姿态控制模块